

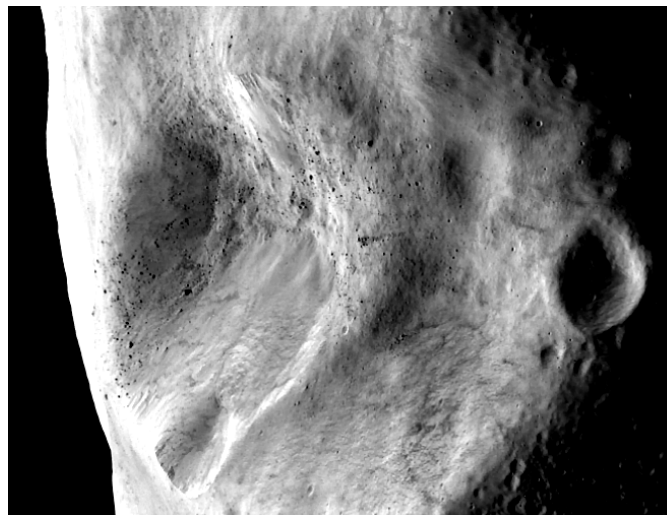
Asteroids

Properties of Asteroids

Asteroid spectra and meteorites

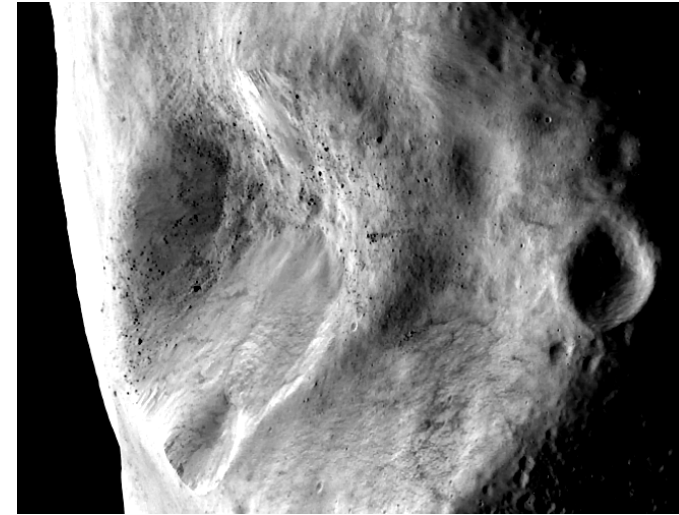
Water on Asteroids

Missions to Asteroids

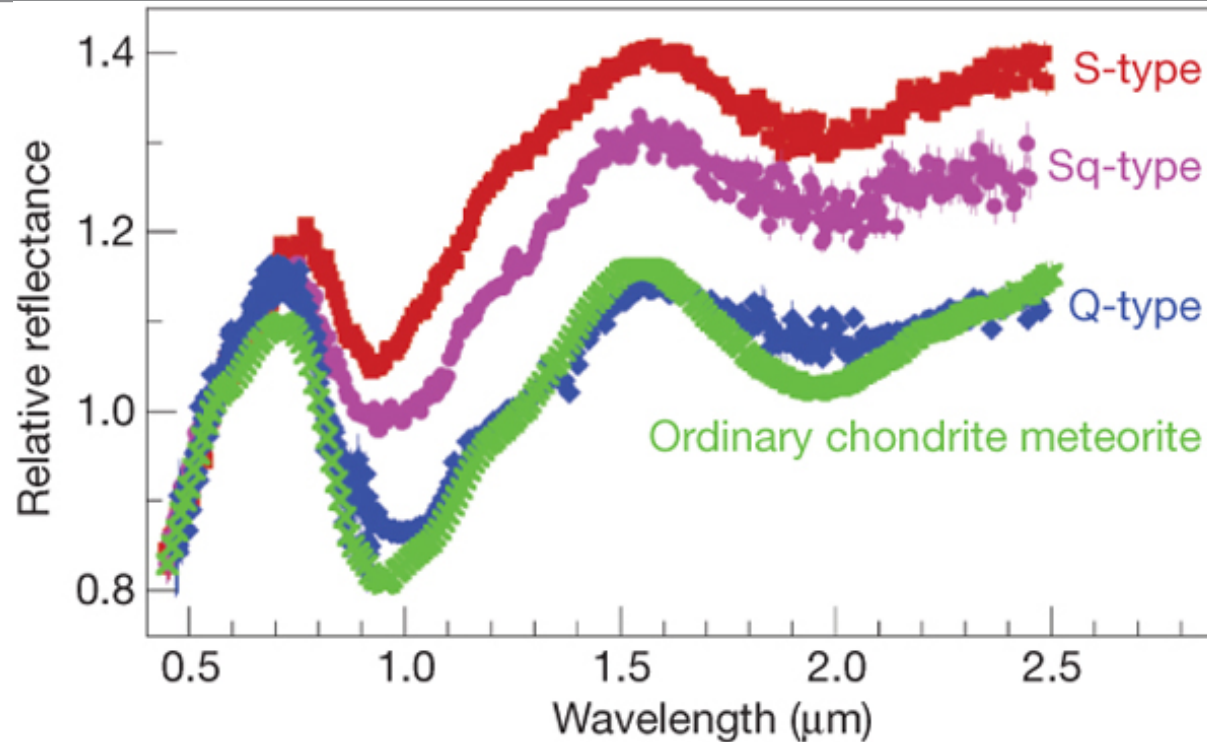


How do asteroids look like?

- Irregularly shaped bodies, covered with impact craters
 - Most of them are fragments from catastrophic impacts or “rubble piles” (re-accumulated material from disruptive impact)
- Composition: Silicate or metallic
 - Many “taxonomic types” with different spectra known
 - Corresponds to different composition

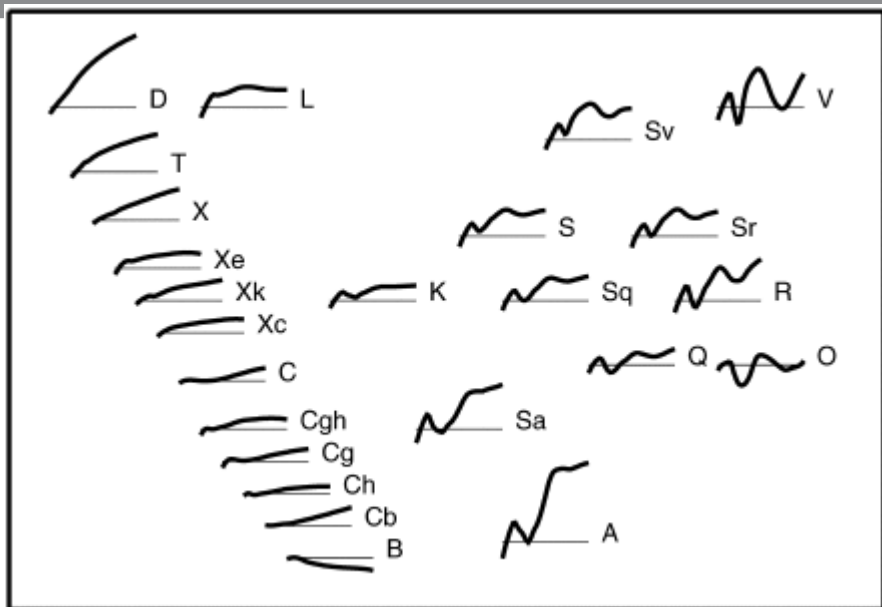


How do we know?

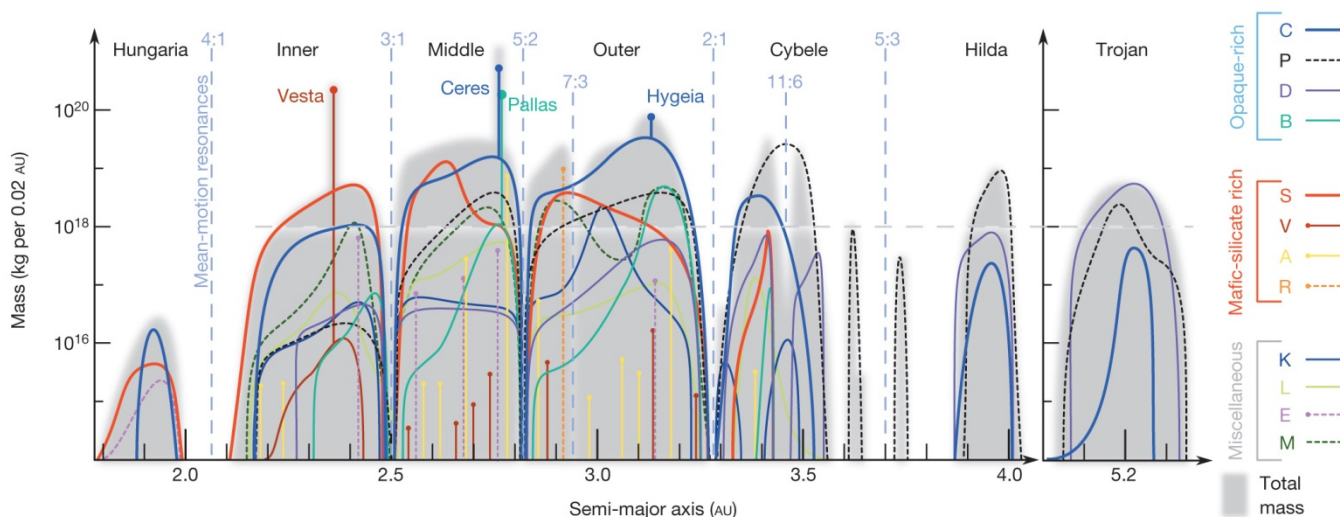


RP Binzel *et al.* *Nature* **463**, 331-334 (2010) doi:10.1038/nature08709

- Meteoritic analogues provide compositional information
- Complication due to space weathering

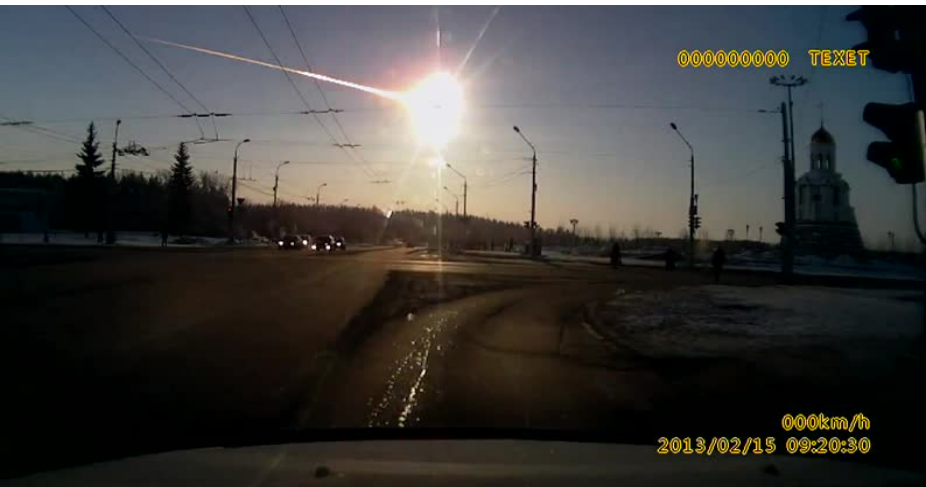


Asteroid classification (de Meo et al. 2009)



- Taxonomic types vs. Position in the solar system

Asteroids: Connection with Meteorites



Explosion of asteroid close to Chelyabinsk, Russia, Feb. 2013



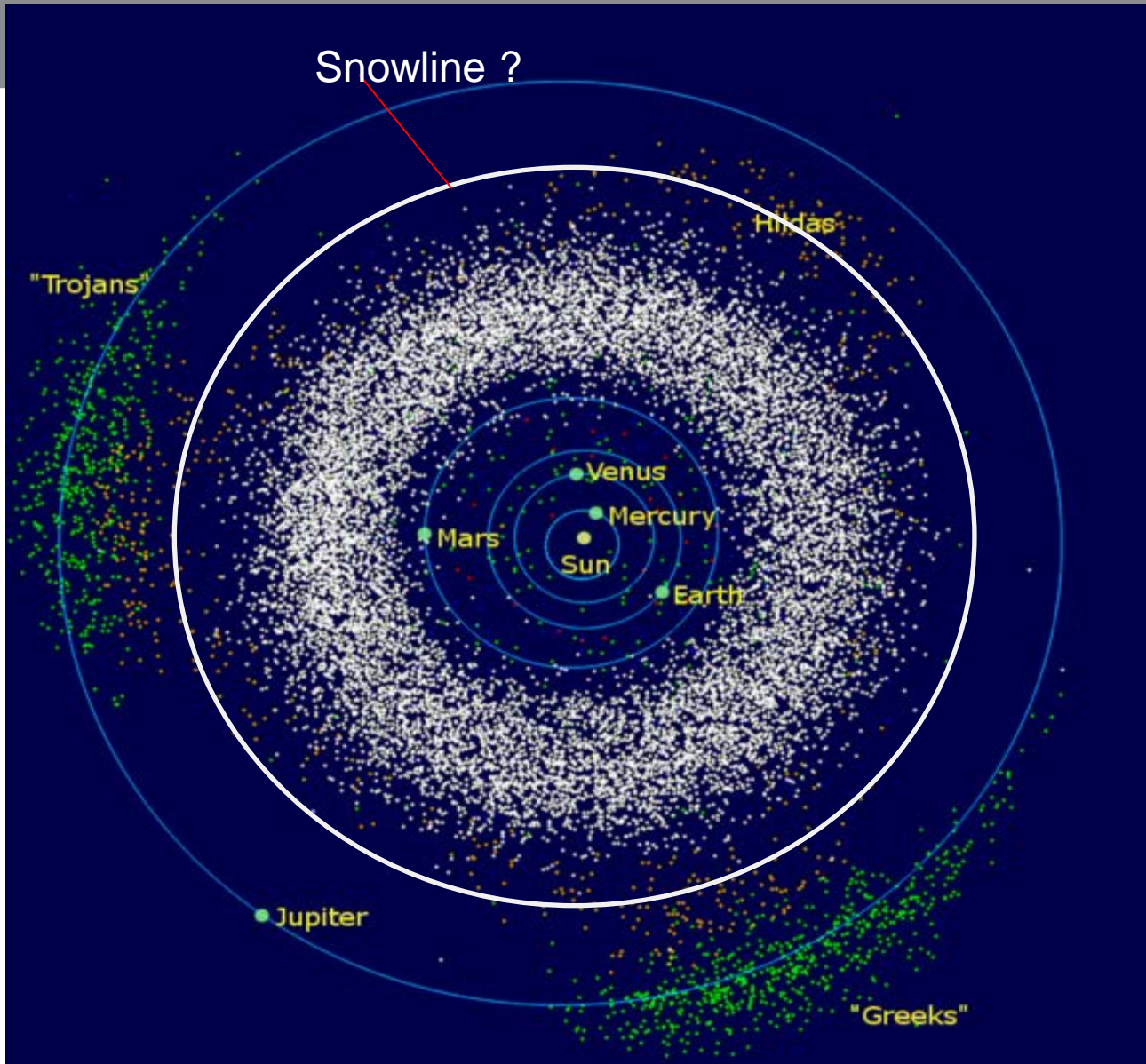
Meteorit found close to Chelyabinsk

Impact Craters



Meteor Crater, Arizona diameter 1200 m, age 50000 years

Volatile rich and inert (rocky) bodies



Classical picture:

Snowline in the solar system somewhere between asteroid main belt and Jupiter's orbit:

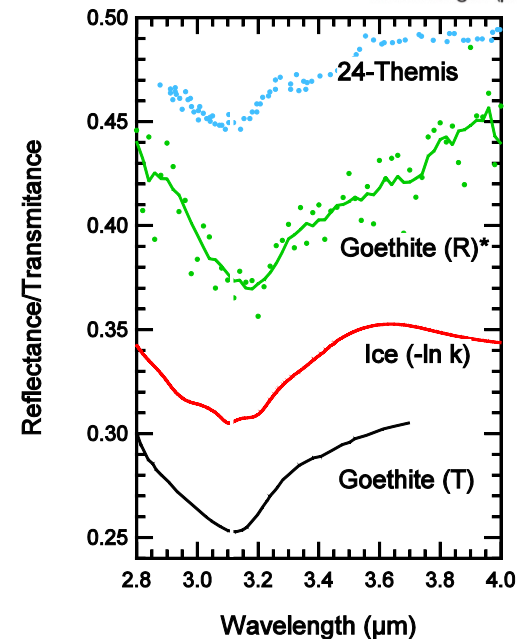
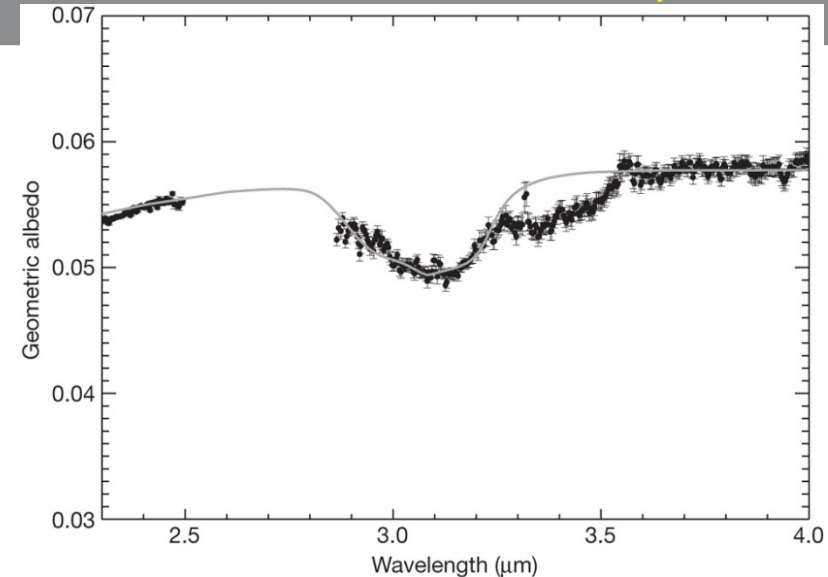
- Giant planets and comets are icy
- Terrestrial planets and asteroids are dry
 - Earth oceans come from outer solar system !?

After: Wikipedia

Transition objects or: Water in the asteroid belt

I Hydrated features in asteroid spectra

- 3 micron absorption in spectra of some asteroids is indicative of hydrated minerals
- It is claimed that the spectrum of 24 Themis cannot be explained by hydrated minerals alone, but requires water ice on the surface (Campins et al. 2010, Rivkin & Emery 2010)
- The mineral goethite (FeO-OH) has been suggested as an alternative to water ice (Beck et al. 2011)

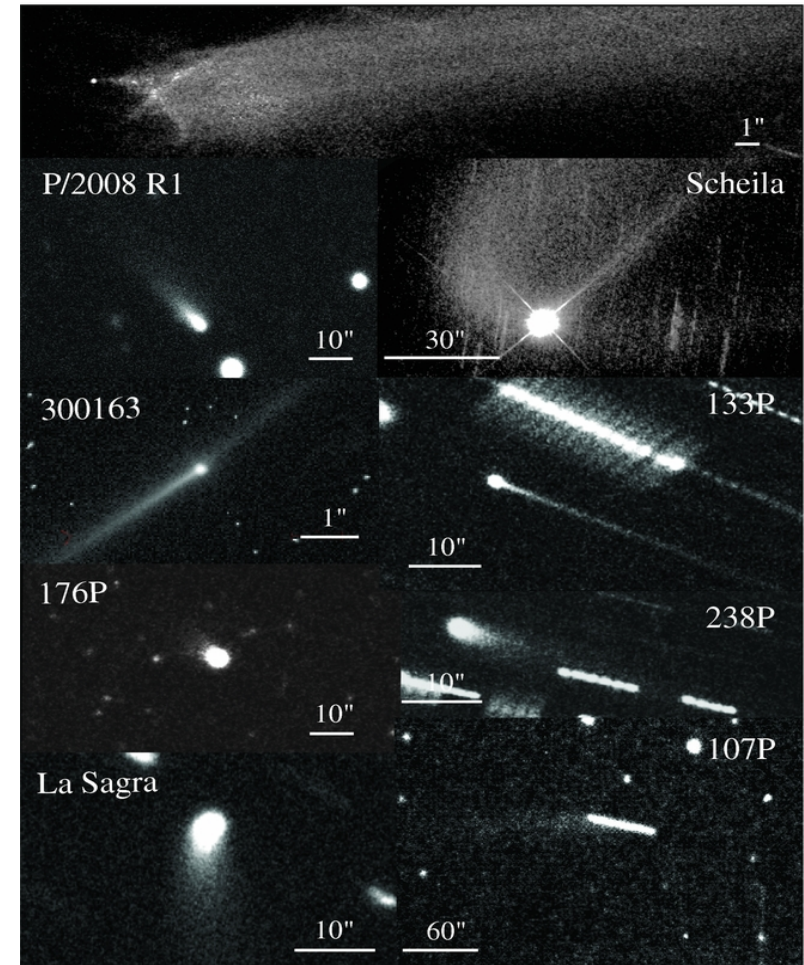


Beck et al. 2011

Water in the asteroid belt?

II Active asteroids or main belt comets

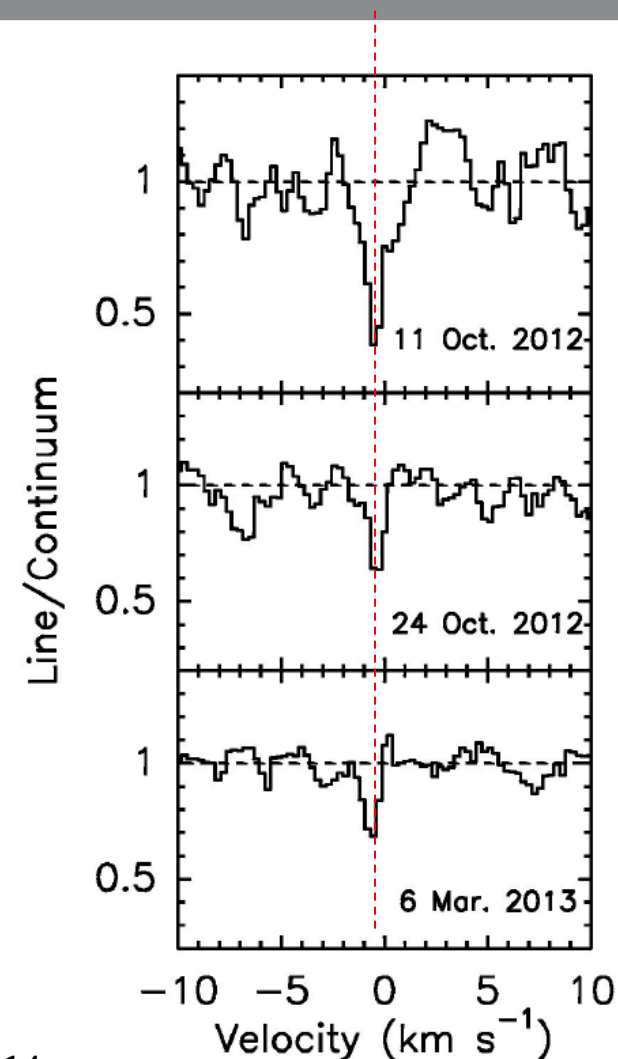
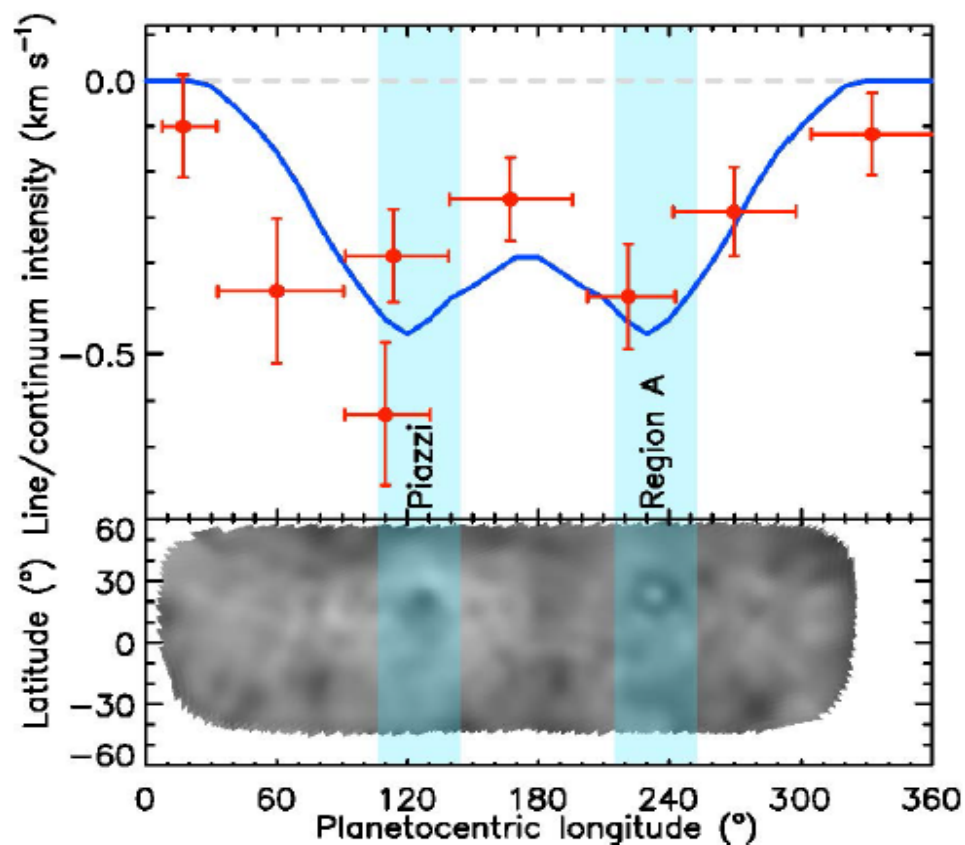
- While some of them are most likely driven by impact or fast rotation, some show recurrent activity around perihelion (133P and 238P)
 - Comet-like activity driven by water sublimation is the most natural explanation
- Several main-belt objects exhibit a dust coma and/or tail
 - No water or other volatiles detected so far



Jewitt 2012

Water detected on Ceres

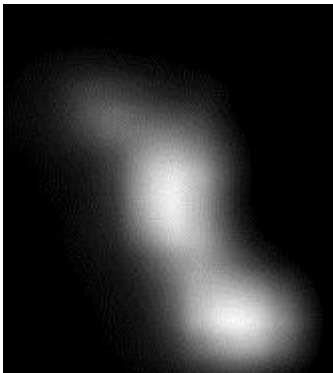
- Water vapour detected with high resolution spectroscopy from Herschel Space Observatory



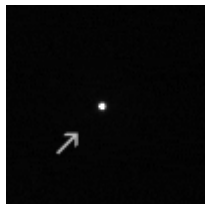
Küppers et al. 2014

- Flybys allow a quick look on individual asteroids
 - Many of them are a secondary target that can be passed by on the way to the main mission target (e.g. Galileo, New Horizons, Rosetta)
 - Allows high resolution imaging and spectroscopy
 - Ten successful asteroid flybys so far, large variety in quality of output data

Braille (DS 1)



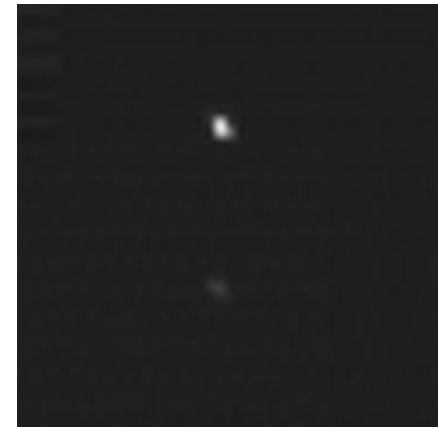
Masursky (Cassini)



AnneFrank (Stardust)



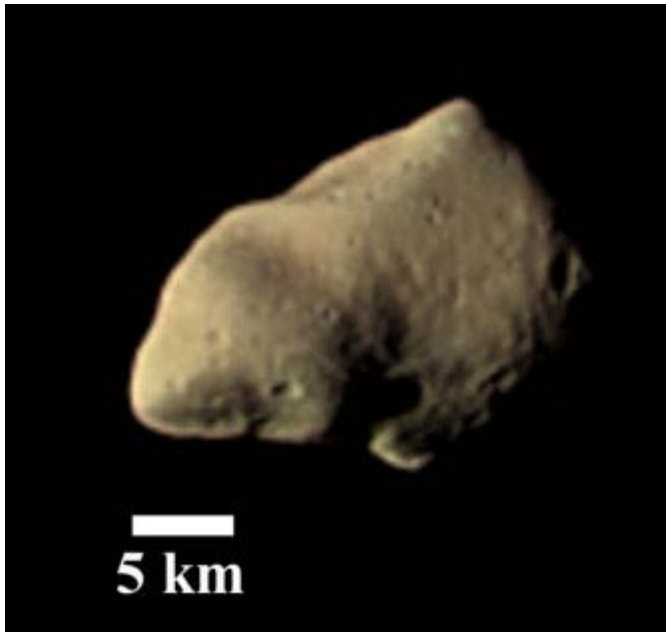
APL (New Horizons)



Source:
Wikipedia

Missions to Asteroids: Flybys (II)

- First Asteroid flybys: Galileo on its way to Jupiter



Source: [Granahan 2011](#), Icarus

Gaspra: First asteroid observed from space (1991)



Source: [NASA/JPL](#)

Ida and Dactyl: First asteroid satellite detected! (1993)

Missions to Asteroids: Flybys (III)

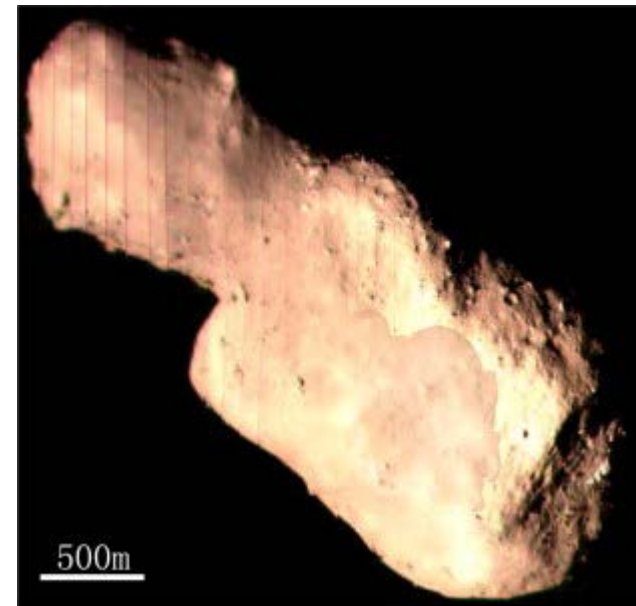
NEAR-Shoemaker flyby of Mathilde



Source: Wikipedia

Huge craters and low density:
Impact in porous bodies causes
compression, not destruction?

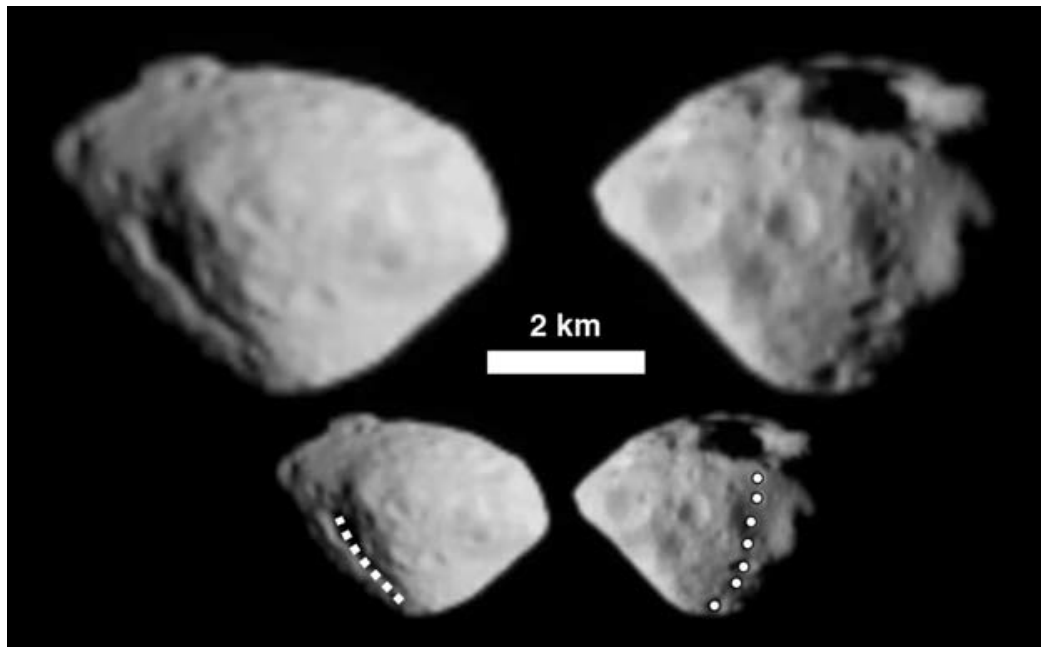
Chang'e 2 flyby of
Toutatis



Source: Zou et al., Icarus 2014

Rosetta flyby of Steins en route to comet Churyumov- Gerasimenko

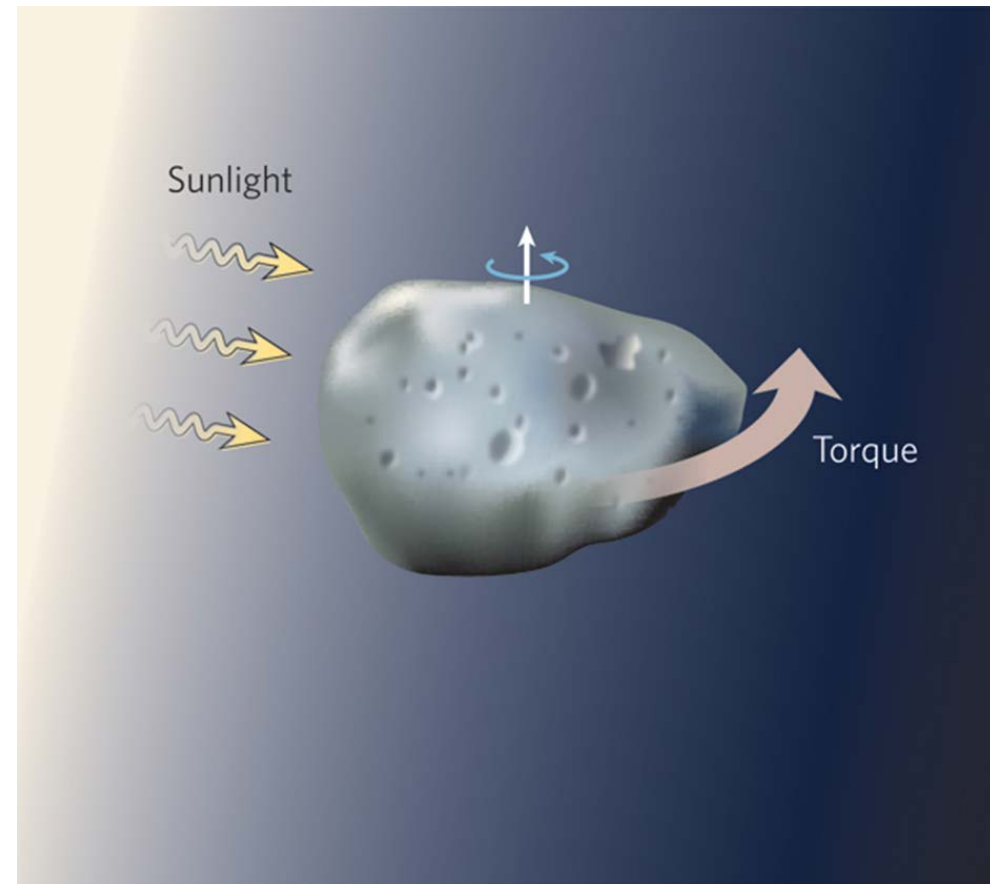
Asteroid Steins



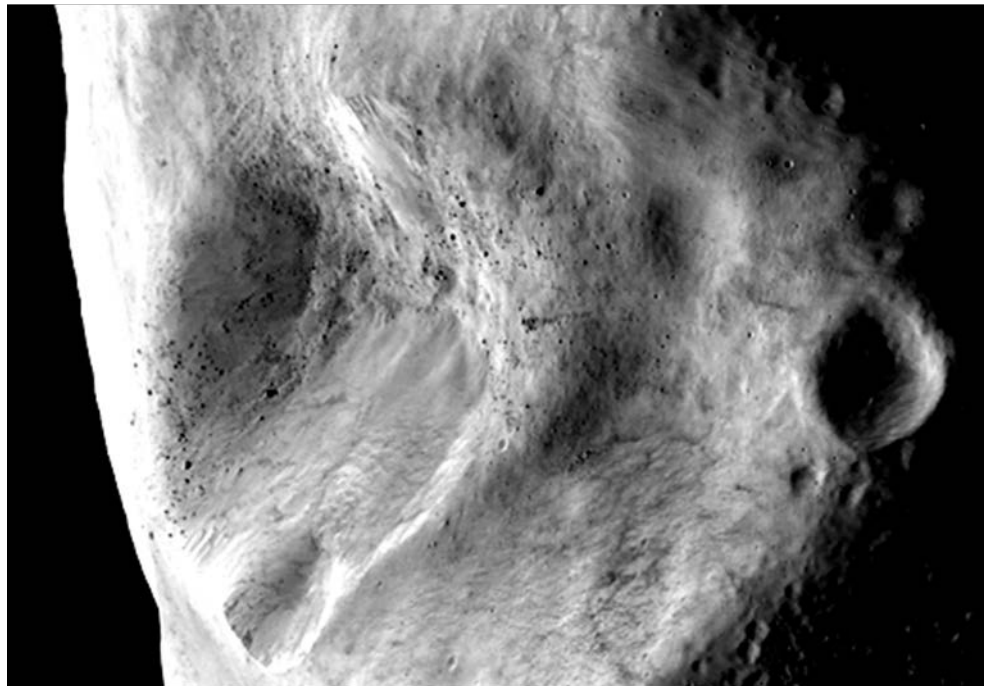
Source: Keller et al. 2010, Science

First spacecraft observation of object shaped by fast rotation («YORP-Effect»)

YORP effect



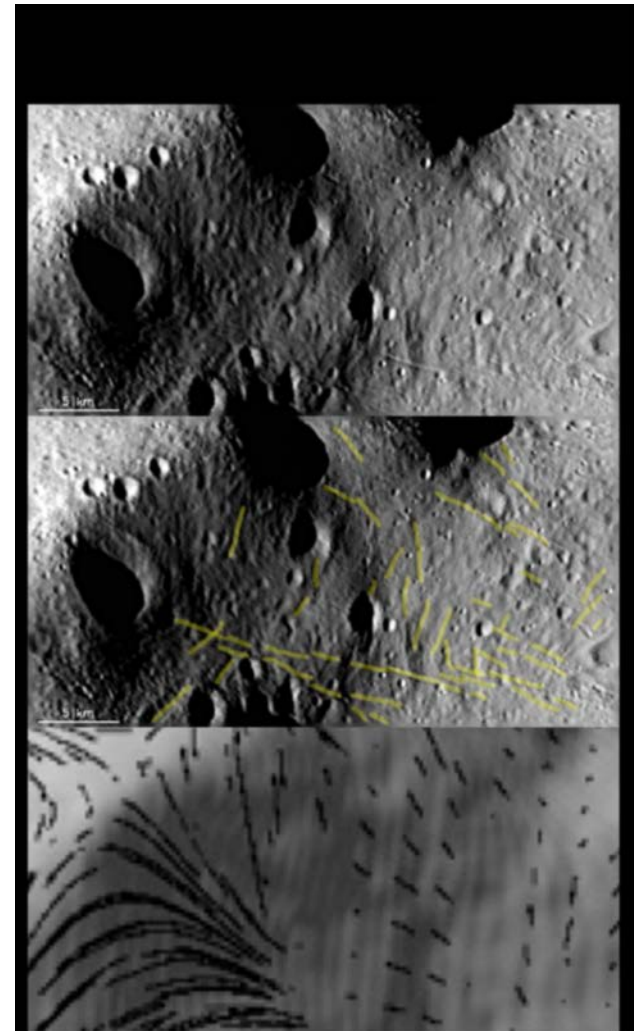
Asteroid Lutetia



Credit: ESA 2010 MPS for OSIRIS Team
MPS/UPD/LAM/IAA/RSSD/INTA/UPM/DASP/IDA

Linear structures explained by waves from big impacts

Example of lineaments and modelled velocity field



Source:
Jutzi et al.
Space Agency

Missions to Asteroids: Rendezvous I



- NEAR-Shoemaker mission Rendezvous with asteroid Eros: First Asteroid Rendezvous
- Orbit insertion failed in 1999, than successful in 2000
 - NEAR stayed at its target for a year
- Mission ended with improvised landing
- First detailed explanation of regolith processes (e.g. boulder formation and destruction)



Source: Wikipedia



Source: NASA/APL

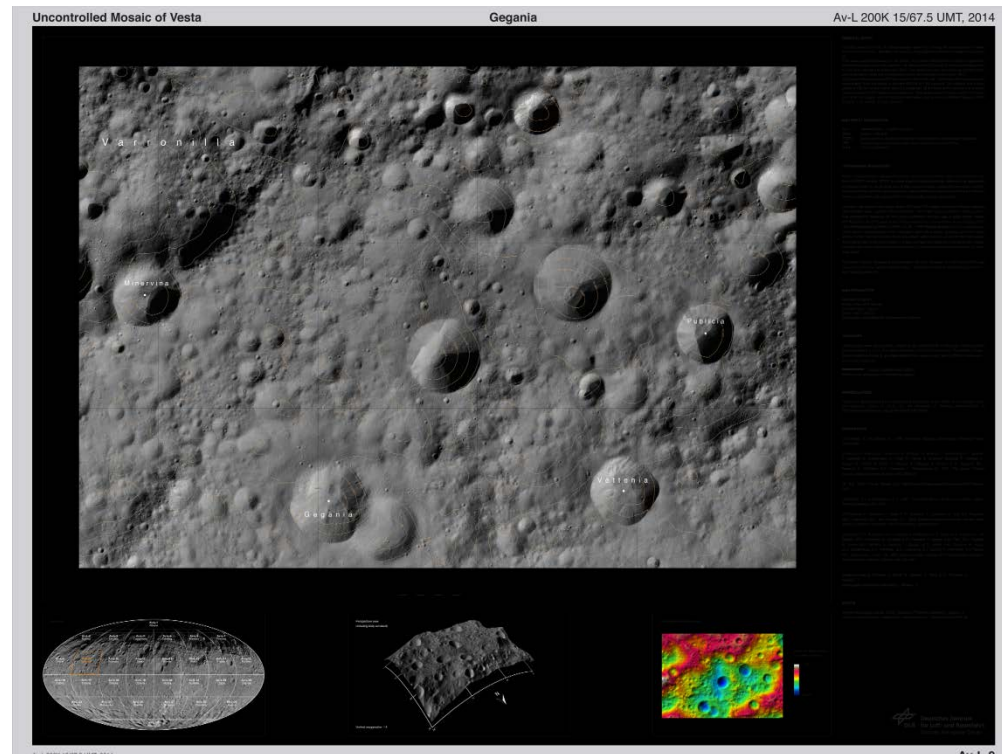
Missions to Asteroids: Rendezvous II

➤ DAWN mission to Vesta and Ceres

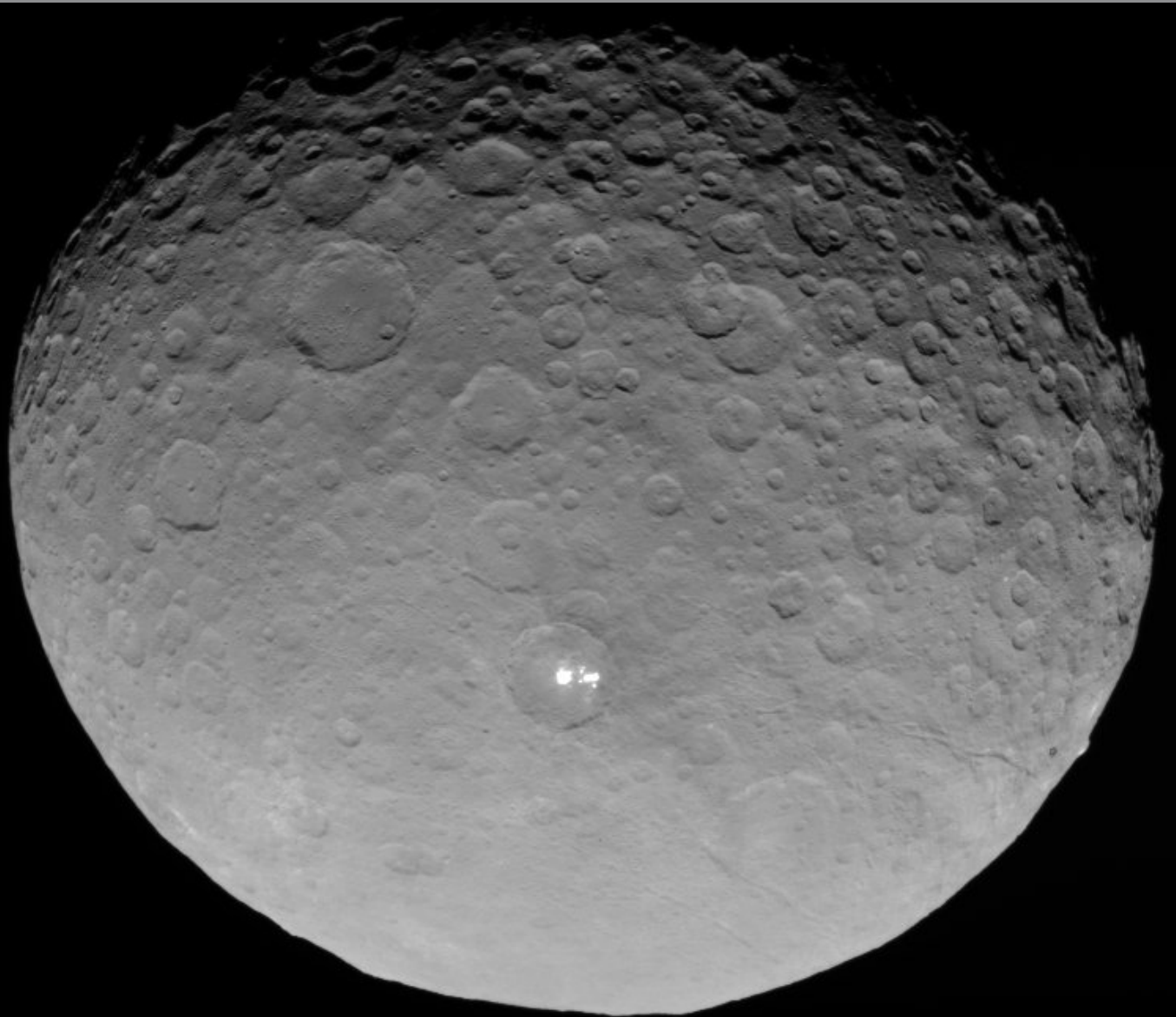
- Visit of largest asteroids (Ceres is actually classified as dwarf planet)
- Orbit around Vesta in 2011
- Rendezvous with dwarf planet Ceres in 2015
- Surface features on Vesta interpreted as transiently flowing water (perhaps)



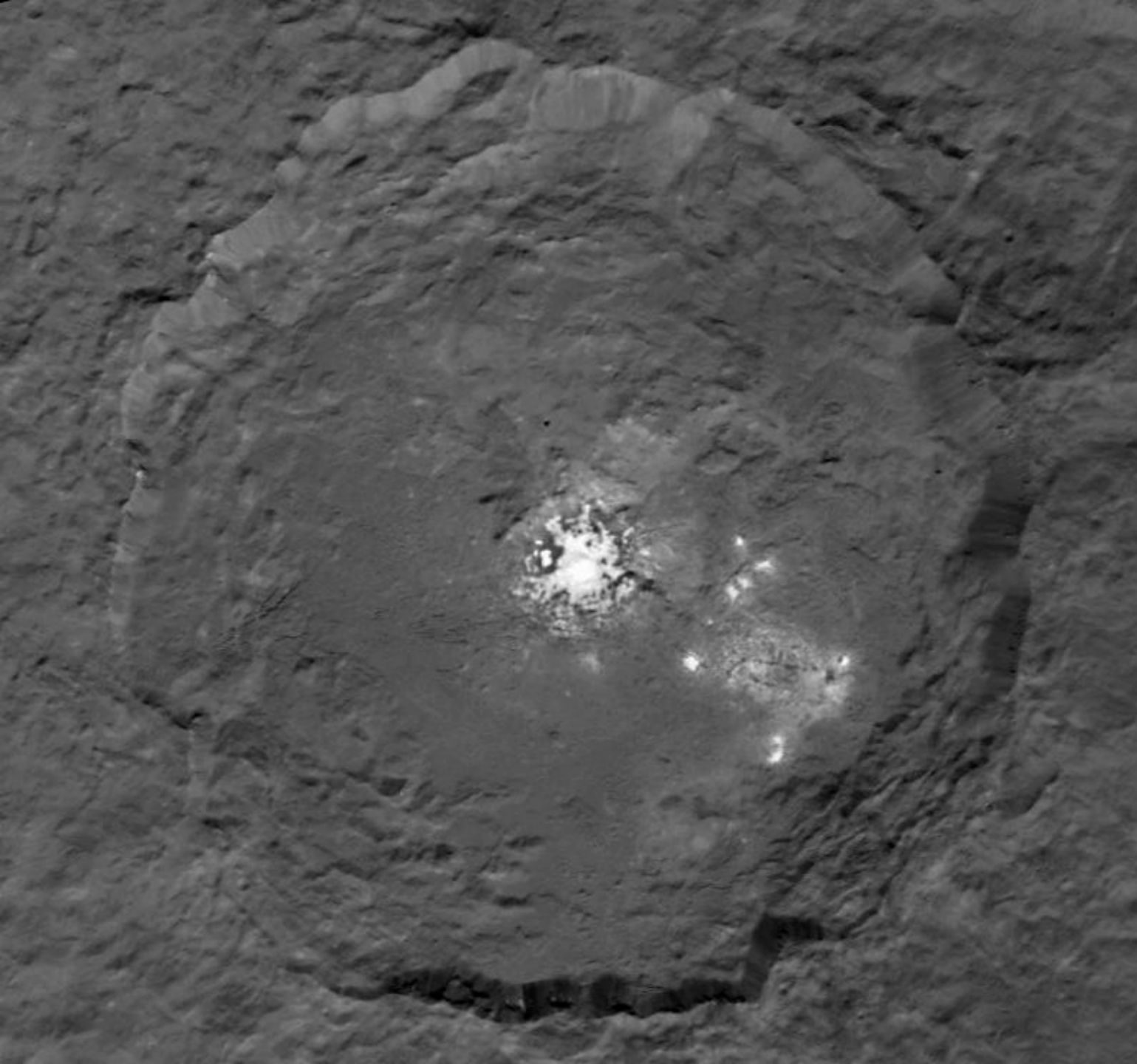
Credit: NASA/JPL-Caltech/UCLA/MPS/DLR/IDA



DAWN at Ceres









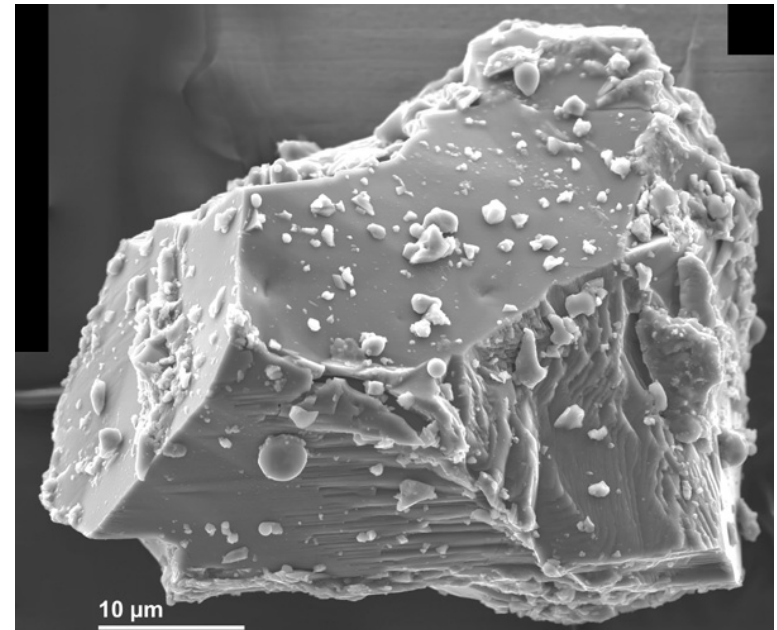
Missions to asteroids: Sample return



- First sample return mission: Hayabusa to Asteroid Itokawa
- Launch in 2003, at Itokawa 2005-2006, return to Earth 2010
- Hayabusa fired a pellet into the asteroid surface and collected ~1500 dust grains (<300 μm , most <10 μm)
 - Firing most likely actually failed!
- Meteorite analog of asteroid confirmed, direct measurement of space weathering



Copyright: JAXA



Credit: Okayama University/JAXA European Space Agency

Missions to asteroids: Future sample return missions: Hayabusa 2

Hayabusa2 Mission Scenario

Launch
FY2014

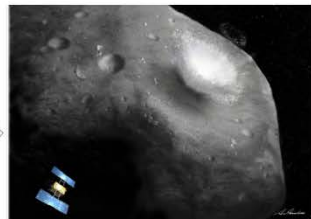


To Asteroid 1999 JU3



Asteroid Arrival
June 2018

Hayabusa2 observes the asteroid by the optical camera, the near infrared spectrometer, Laser altimeter, etc. Then it approaches near the surface of the asteroid, releases small rovers (MINERVA2) and a small lander (MASCOT), and tries to get the surface material.



The impactor will collide to the surface of the asteroid to create an artificial crater. The crater will be observed to study underground matter.

Spacecraft will go some other places such as Lagrange point.



Asteroid Departure
December 2019

Earth Return
December 2020

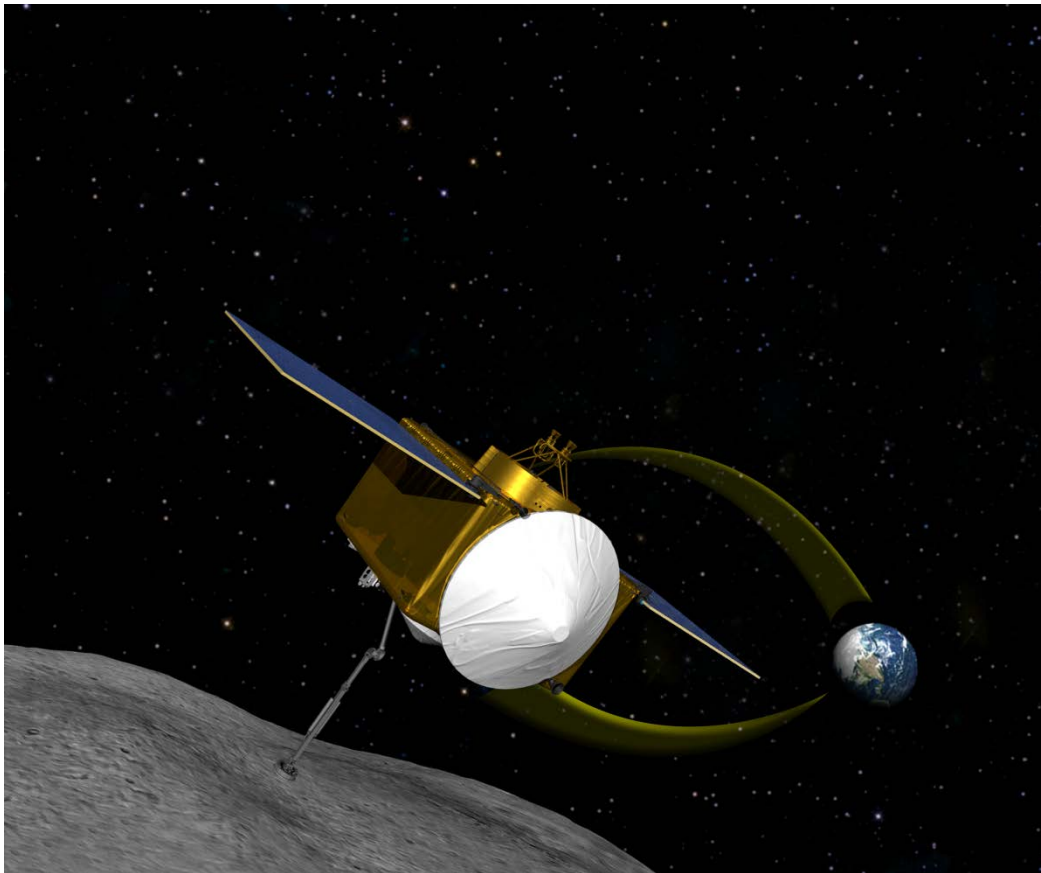
Sample Analysis

After confirming the safety, touchdown will be performed to get the subsurface matters.

Missions to asteroids: Future sample return missions: OSIRIS-Rex



- Launched October 2016, return 2023
- Sampling with an robotic arm, up to 2 kg of material



Missions to Asteroids: Asteroid mining?

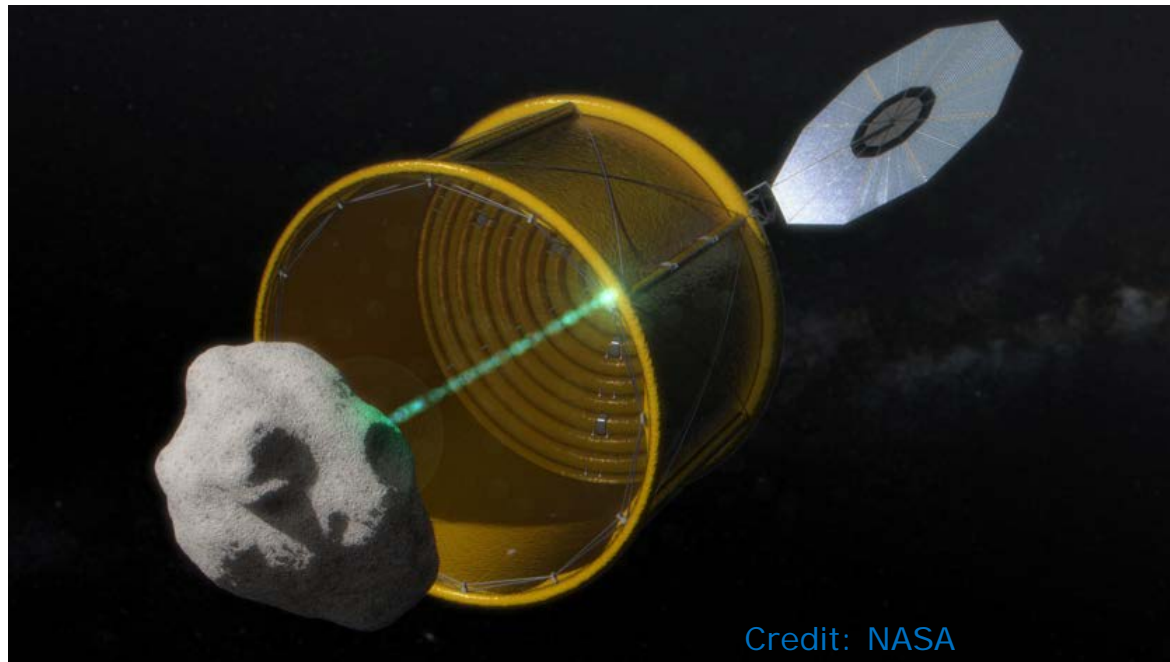
- Idea of asteroid mining became popular with «Planetary Resources» project
 - Idea is to return precious elements that are rare on Earth
- Economic feasibility?
 - See price per kg of sample return missions
- Maybe a future option for in space utilisation



➤ Asteroid Redirect Mission

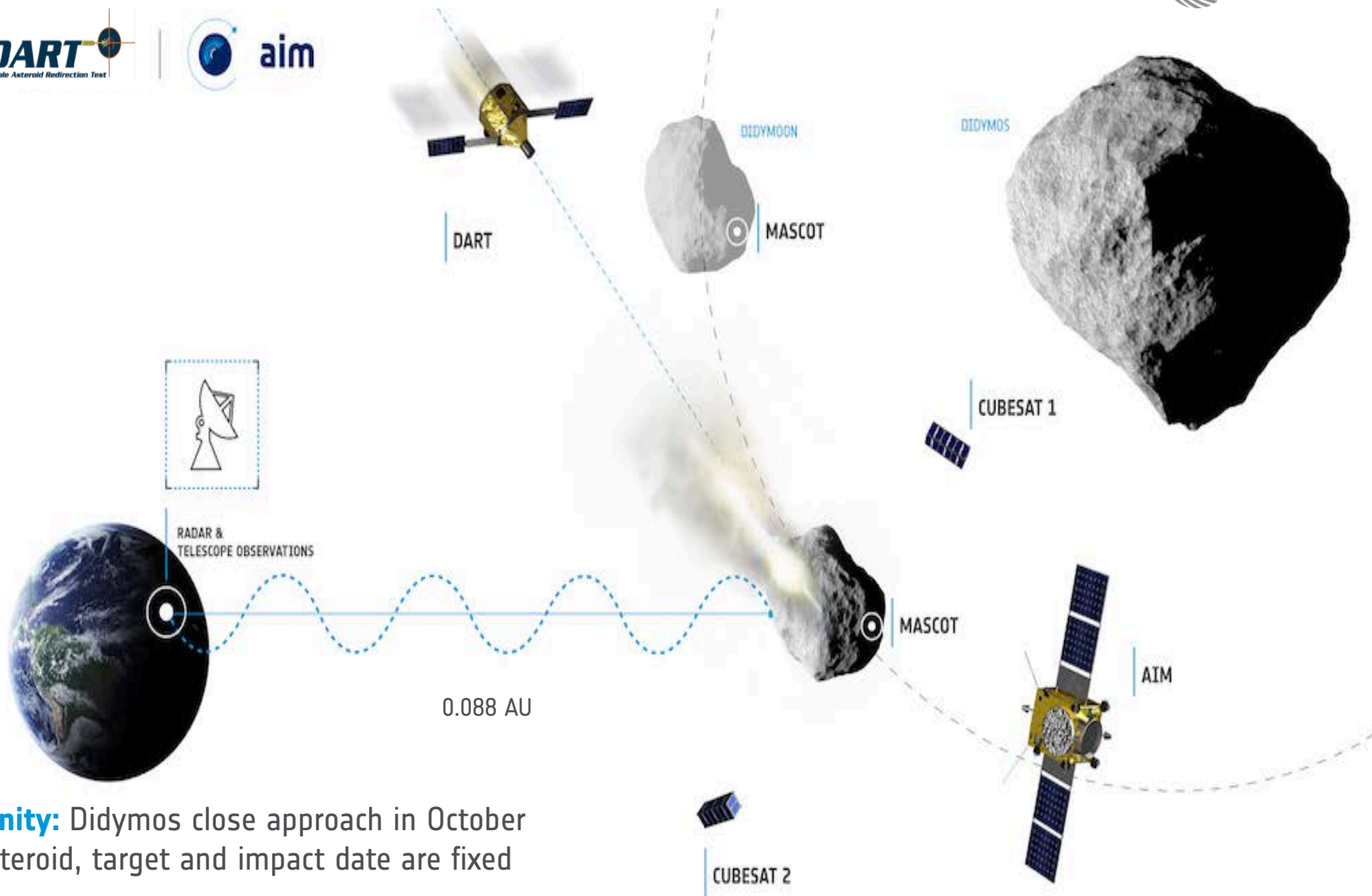
-> See presentation by Humberto Campins

Original concept



Credit: NASA

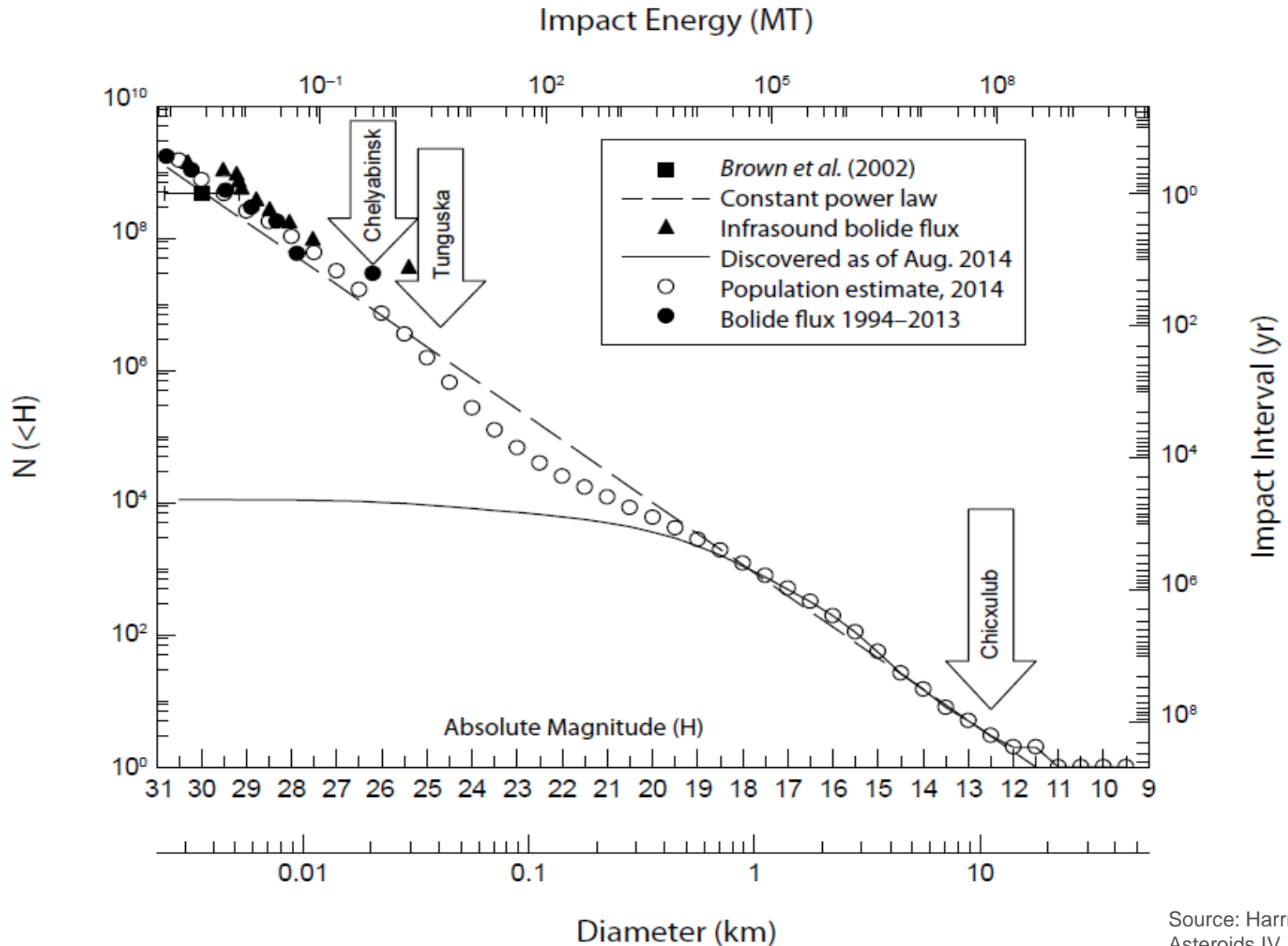
AIDA COOPERATION: Asteroid deflection



→ **opportunity:** Didymos close approach in October 2022 asteroid, target and impact date are fixed

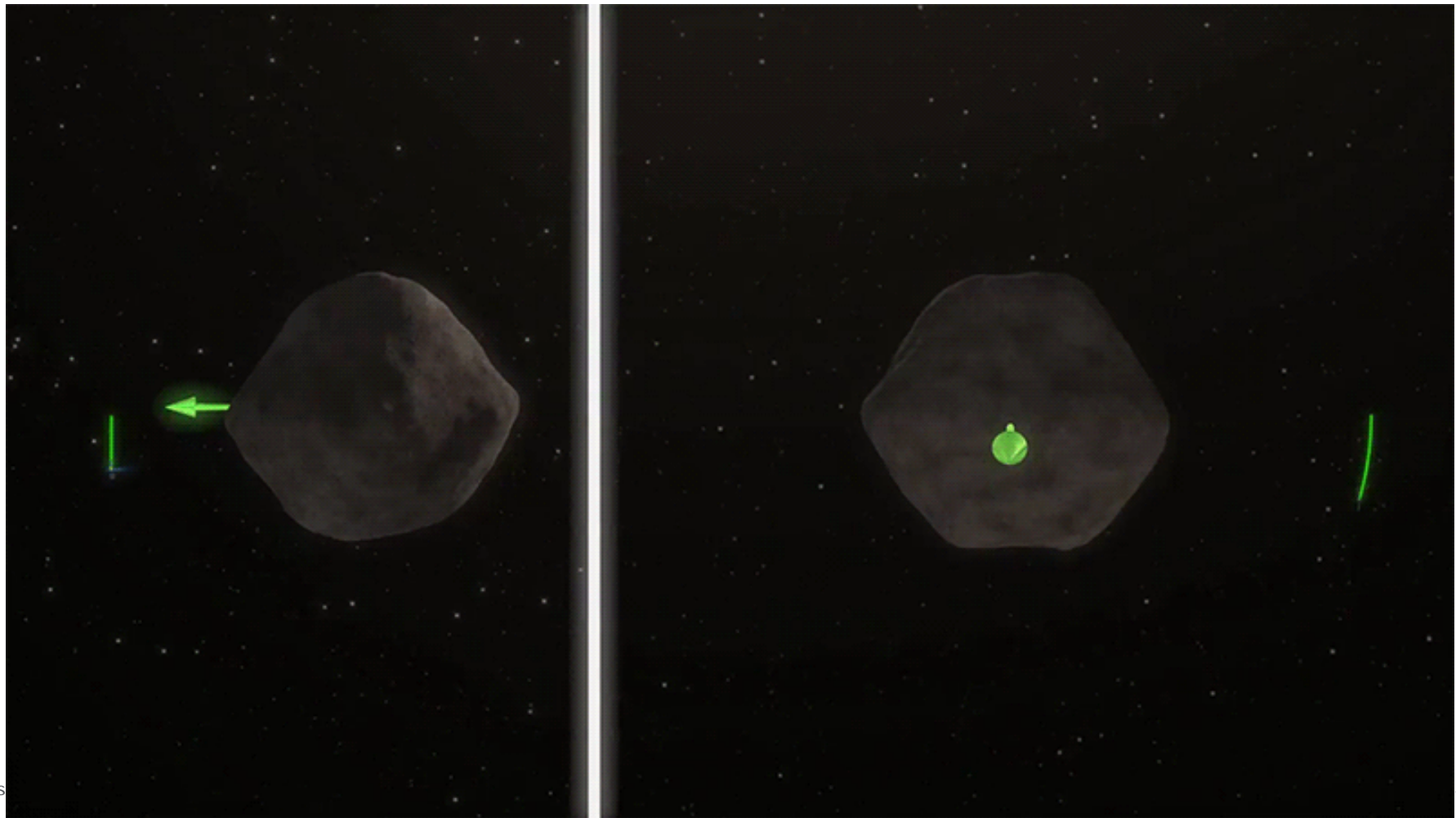


Asteroid impacts: What is the danger?



Methods of asteroid deflection

I Gravity tractor: Gentle, safe, but not very efficient



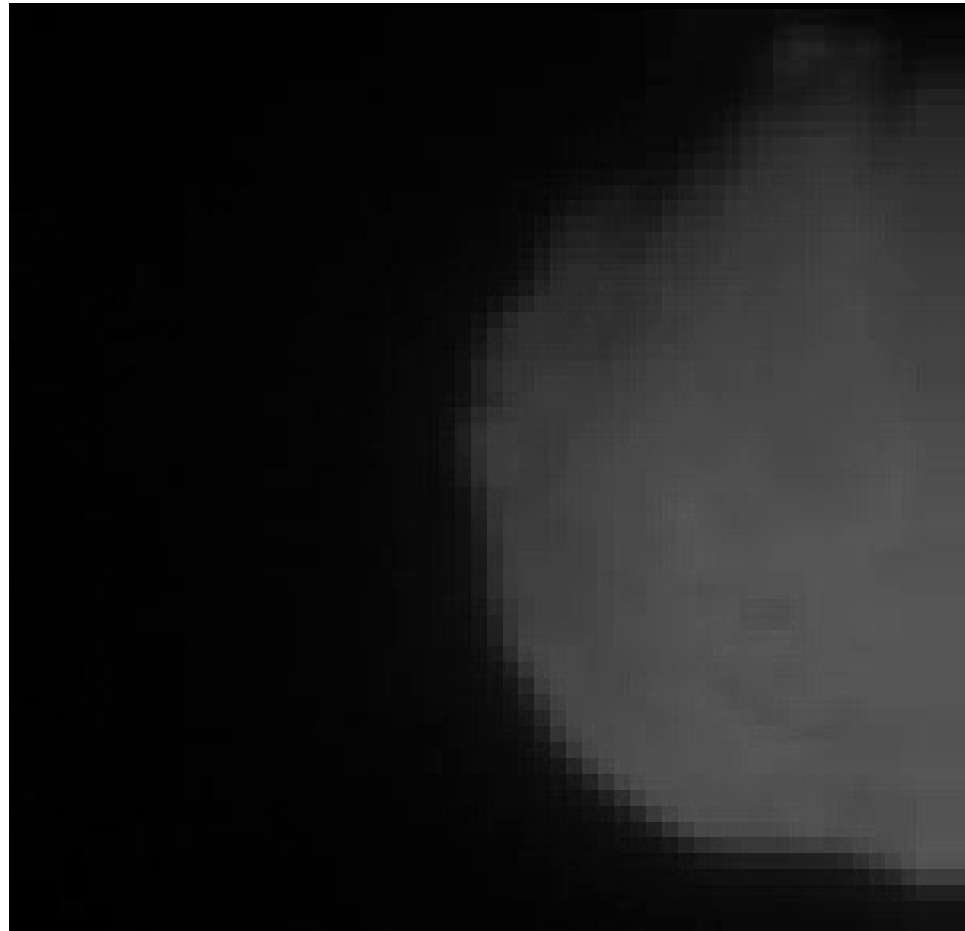
Methods of asteroid deflection



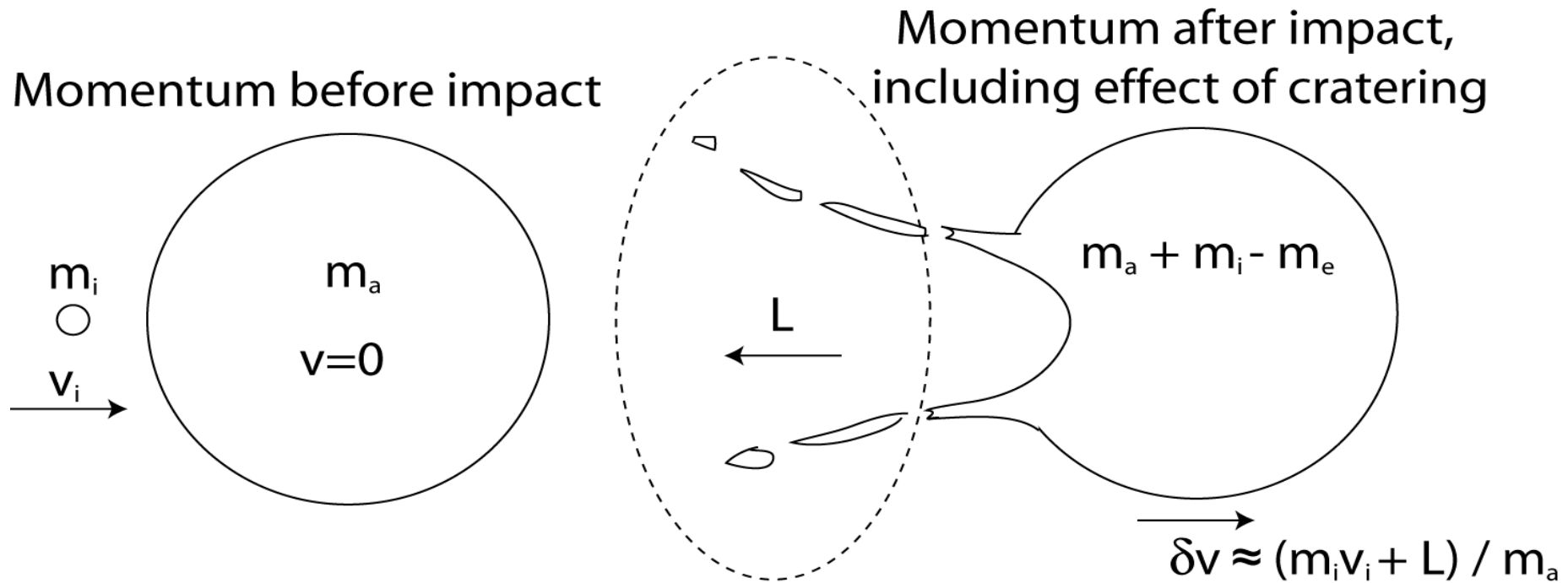
II Nuke: Efficient, but highly dangerous



III Kinetic impactor: intermediate efficiency, technically most advanced



Efficiency of an impact



ASTEROID IMPACT MISSION (AIM)



aim

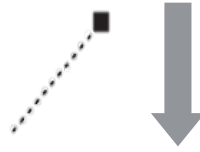
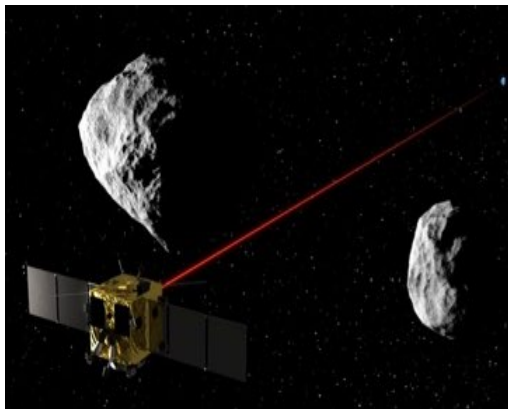


- Technology demonstration mission
 - Fast turnaround allows test of technologies being developed for later missions
- Science is “ride-along”



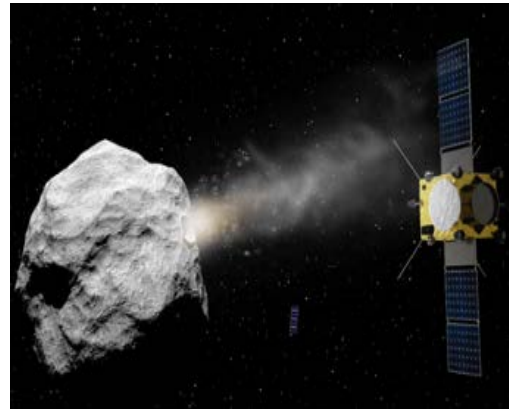
TECHNOLOGY

DEMONSTRATION



ASTEROID IMPACT

MITIGATION

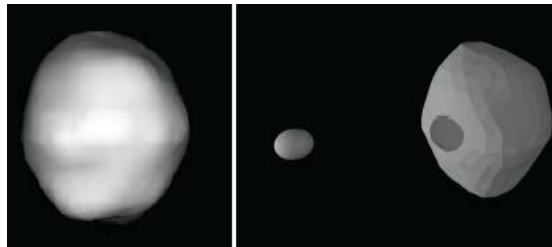
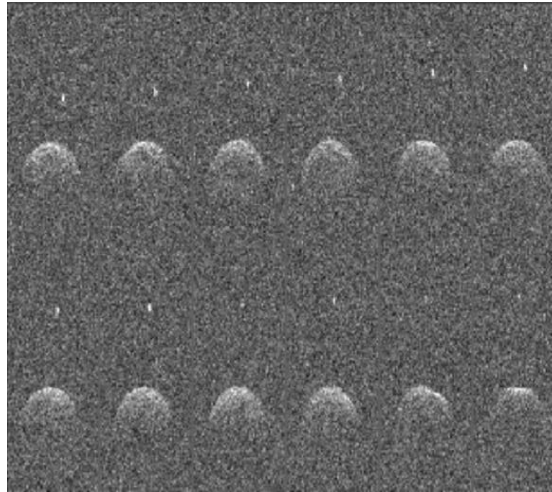


SCIENCE



DIDYMOS: A PERFECT TARGET

- Asteroid observed by ground telescopes and radars
- Heliocentric orbit well known
- Shape and size of primary well known (not Didymoon)
- Orbit plane orientation to be confirmed in 1Q 2017
- Didymoon
- Size representative of a potentially hazardous object (generating casualties independently from impact location on Earth)
- Mission takes advantage of close flyby of earth in 2022 (0.07 AU)



Chelyabinsk meteor (Feb 2013): 1500 injuries, 7200 damaged buildings



AIM TECHNOLOGY RESEARCH OBJECTIVES



T#1	Demonstrate deep-space inter-satellite communication network	<ul style="list-style-type: none"> <input type="checkbox"/> Deploy up to two 3U cubesats (or any combination of units) <input type="checkbox"/> Demonstrate inter-satellite link network between AIM, COPINS and MASCOT-2
T#2	Demonstrate asteroid landing and extended operations on Didymoon	<ul style="list-style-type: none"> <input type="checkbox"/> Demonstrate landing on small (170 m) asteroid and inter-satellite link in deep-space <input type="checkbox"/> Test long-lived payload operation i.e. transmission radar, surface imaging, radiometry.
T#3	Infra-Red Instrument Navigation Aid <i>(IRINA)</i>	<ul style="list-style-type: none"> <input type="checkbox"/> To demonstrate the use of an IR instrument potential to support rendezvous phases
T#4	Qualify an end-to-end 2-way deep-space optical communications system	<ul style="list-style-type: none"> <input type="checkbox"/> Primary goal: transmit full asteroid 1m resolution map before DART arrival (goal, transmit images of the impact)



AIM (alone) PRIMARY SCIENCE OBJECTIVES

S#1 Didymoon size, mass, shape, density	Mass => momentum size => shape, volume, gravity density => internal structure	Camera (VIS), LIDAR (OPTEL-D), radio tracking
S#2 Didymoon dynamical state	Momentum transfer Indirect constraints on interior structure	VIS
S#3 Geophysical surface properties, topology, shallow subsurface	Composition, mechanical properties, thermal inertia => Interpretation of impact	VIS, Thermal Infrared Imager (TIRI), High Frequency Radar (HFR), Accelerometer on MASCOT
S#4 Deep-internal structure of the moonlet	Interpretation of the impact Origin of binarity	Low Frequency Radar (LFR)
S#5 Optical, IR, Radar calibration	Simultaneous ground and space-based measurements to calibrate ground-based observations and extrapolate to other objects observed from the ground.	VIS, TIRI, HFR



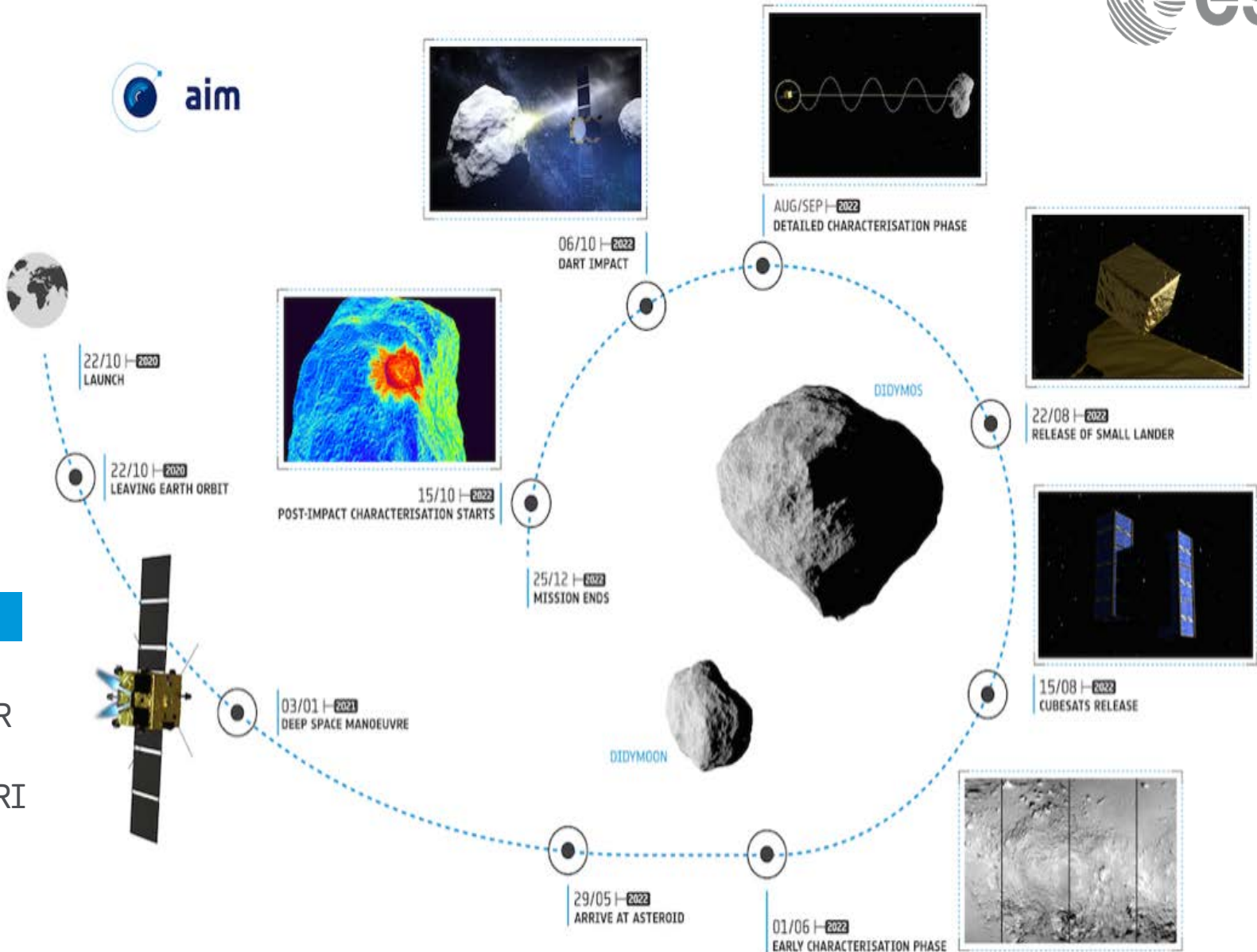
AIM (with DART) SECONDARY SCIENCE OBJECTIVES



S#6 Didymoon post-impact characterisation	Changes due to impact	All
S#7 Didymain characterisation	Origin of the system	VIS, TIRI, HFR, LFR
S#8 Impact ejecta	Properties of ejected dust	VIS, TIRI, HFR
S#9 Ambient dust	Dust in Didymos environment	VIS, TIRI, HFR
S#10 Chemical and mineralogical composition	Asteroid classification, origin of the system	VIS (TBC), TIRI, MASCOT-2 lander



MISSION SCENARIO

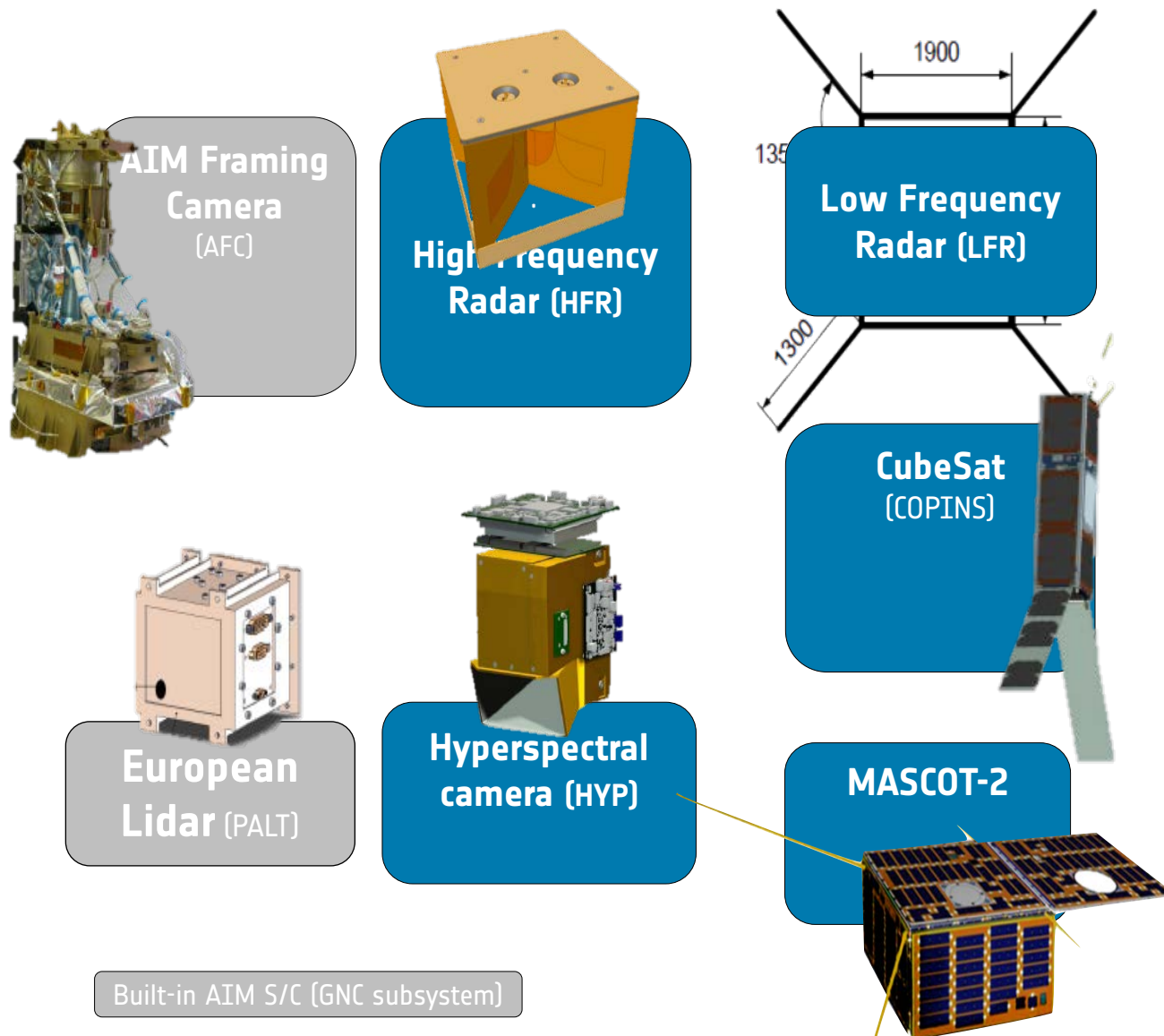


→ PAYLOAD

- VIS
- Mascot-2 + LFR
- HFR
- Hyperspec./TIRI
- Optel-μ
- CubeSats



AIM REFERENCE PAYLOAD



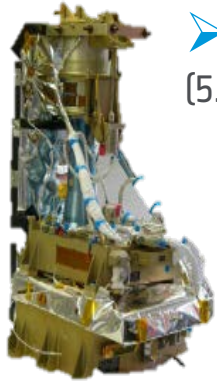
- Several options studied in detail to prepare for proper **interfaces** and **proximity operations**.
- Some payload expected to be committed when mission is approved in Dec.
- Announcement for payload opportunities to be released in **Jan 2017** for any remaining available payload

Options

Optel-D
Optical comms Terminal

TIR Imager
(TIRI)

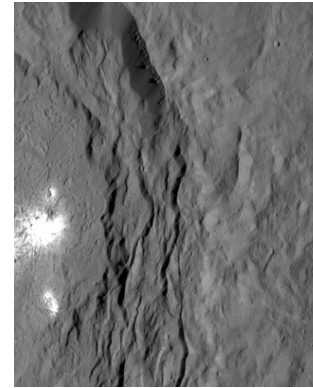
AIM Framing Cameras (AFC), Hyperspectral Imager



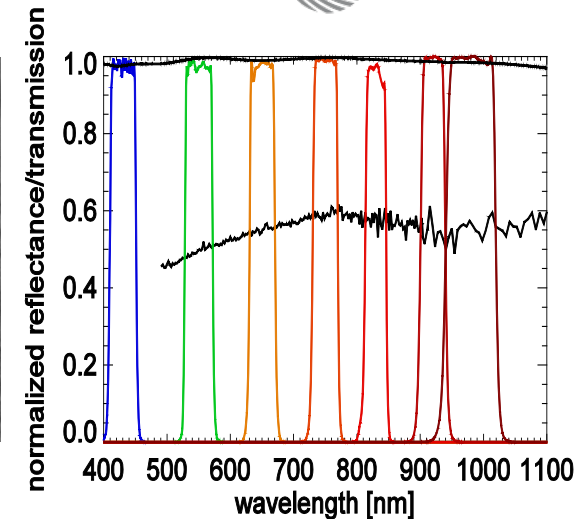
➤ Flight Spares of the DAWN cameras

(5.5° FOV, 93.7 μ rad/pixel, 400-1000 nm, 7 filters)

- spacecraft GNC system, provided by MPI for solar system research
- Used for spacecraft navigation but also science
- Navigation currently being tested at GMV with QM

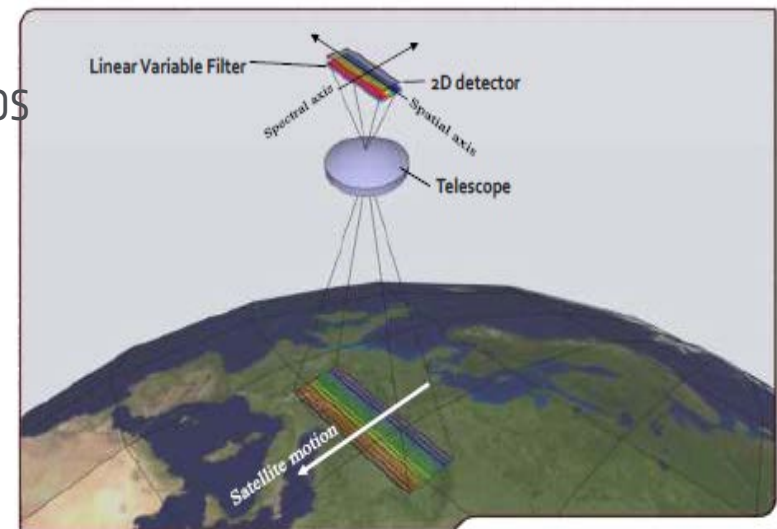


DAWN FC image of Ceres



Compact Hyperspectral imager

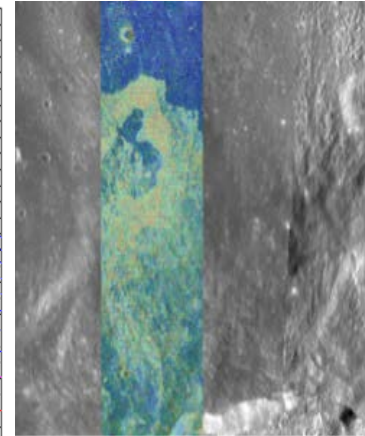
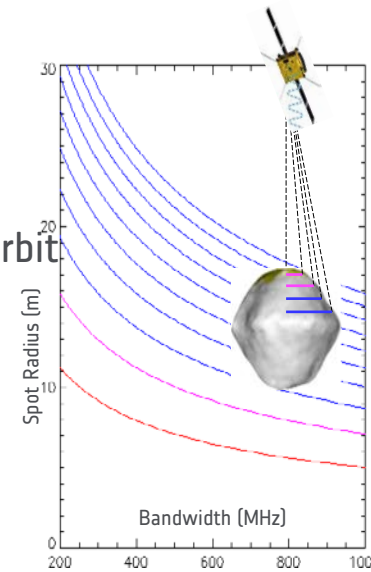
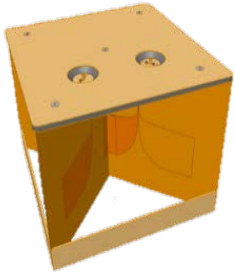
- Grating spectrometer or linear filter fixed on CMOS detector
- Large detector, 7 x 9 deg. FOV at 8 arcsec/pix
- Spectral resolution 5-10 nm
- Wavelength range 470-950 nm
- Developed for Earth observation



Stepped high-frequency frequency radar

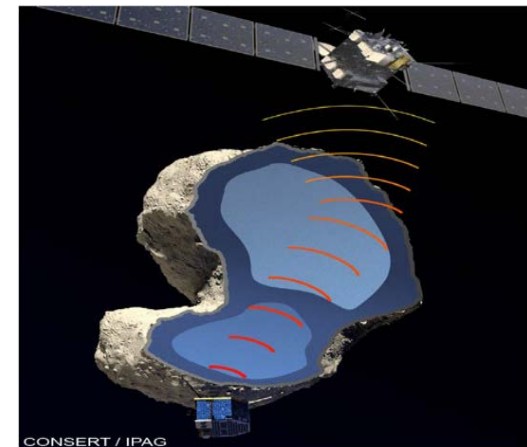
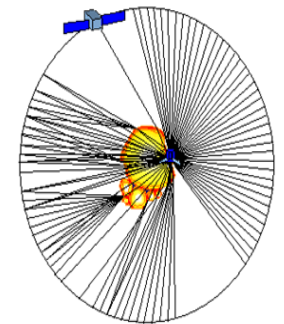
(300MHz to 2.4GHz, 108W power, 2.86kg, 37 x 37 x 27 cm³)

- ◆ determine structure and layering of shallow sub-surface
- ◆ support asteroid mass determination, shape modelling and orbit characterisation
- ◆ observe ejecta cloud
- ◆ support ground-based bi-static radar measurements Arecibo, Goldstone, SRT



Instrument design based on CONSERT (Rosetta)

- Spare components available and TRL6
- Radar type: Bistatic radar (between AIM and MASCOT-2)
- Carrier frequency: 60 MHz
- Bistatic operation through the secondary asteroid



IMAGER (TIRI), MASCOT-2 μ LANDER



➤ TIRI strawman design

- Heritage: MERTIS (Bepi-Colombo), MAIR, HIBRIS, AMS
- Temperature range: 200 K – 450 K
- Spectral range: 8 μ m – 13 μ m (spectral resolution 0.3 μ m)
- Spatial resolution (goal): 2 m @ 10 km
- Field of view: ~5 deg., similar to cameras
- Thermal and physical surface properties

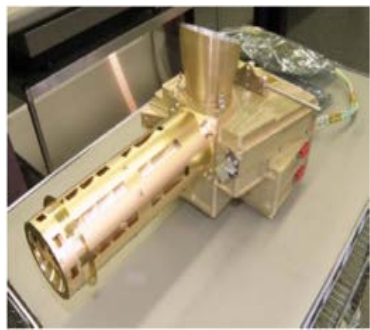
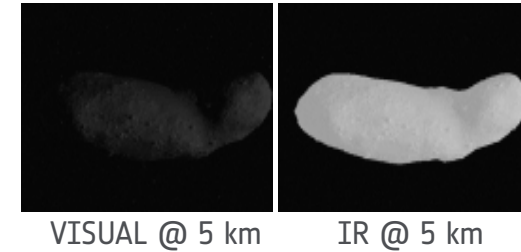
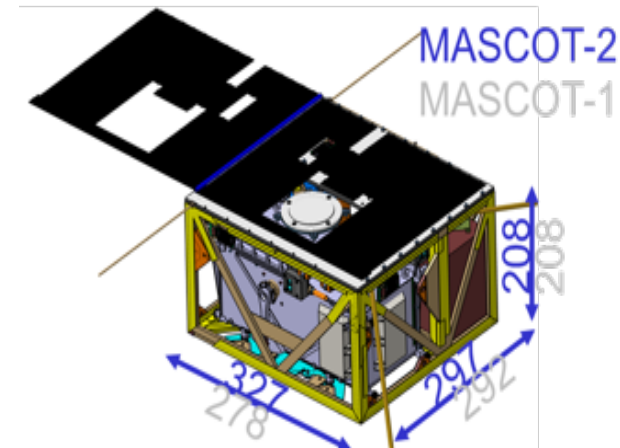


Figure: MERTIS

MASCOT-2 μ lander

- ◆ Development based on MASCOT-1 currently on JAXA's Hayabusa-2 mission
- ◆ Size: 33 x 30 x 21 cm
- ◆ Mass: 15 kg
- ◆ Deployable solar generator cover (supports orientation)
- ◆ 3 months operational lifetime
- ◆ Carries: μ -camera (CAM), low-frequency radar (LFR), radiometer (MARA), accelerometer (DACC)



COPINS: CUBESATS IN DEEP SPACE



ASPECT



- Vis-NIR imaging spectrometer
- Space Weathering
- Shock experiment
- Plume Observations

AGEX



- Mechanical properties of surface material
- Seismic properties of sub-surface
- Determine kinematics prior and after DART

PALS



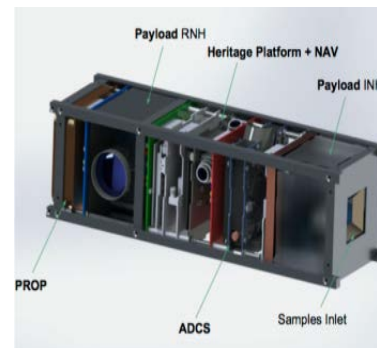
- Characterize magnetization
- Composition of volatiles
- Volatiles released from DART impact
- Super-resolution imaging
- DART collision and plume observation

CUBATA



- Gravity field
- Observe DART impact
- Perform seismology
- Velocity field of the ejecta

DUSTCUBE



- Dust properties with Nephelometer
- Mineralogical composition
- Compliment com demo
- Reflectance of the asteroid surface



- Distribution of asteroids today may show traces of migration in the solar system
- Water is present in the asteroid belt
- Asteroid deflection mission may be the next European Small body mission after Rosetta
 - Decision will be made 2nd Dec.