



Solar Orbiter

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The science



- How does the Sun create and control the heliosphere and how solar activity changes with time?
 - What drives the solar wind and where does the coronal magnetic field originate?
 - How do solar transients drive heliospheric variability?
 - How do solar eruptions produce energetic particle radiation that fills the heliosphere?
 - How does the solar dynamo work and drive connections between the Sun and the heliosphere







- Launch: October 2018
- Cruise phase: 2.3 yr
- Nominal mission: 3.5 yr



- Extended mission: 2.5 yr
- Orbit: 0.28 0.91 AU (Period 150 -180 days)
- Out-of-ecliptic view:
 - > 24° during nominal mission
 - > 32° during extended mission
- Reduced relative rotation
 - Observation of evolving structures for almost a complete solar rotation
- Remote sensing instrument windows
 - Three windows of 10 day each
 - Minimum distance
 - Maximum and minimum heliolatitude







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The spacecraft

solar orbiter

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The spacecraft

- System-level CDR successfully passed
- STM tests completed
- S/C Engineering Test Bench: most of the instrument SF and FFTs completed
- Majority of FM units delivered, including the core structure
- Integration started
- Ground segment
 - Design review of Science Ground Segment completed
- Launch vehicle

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- NASA will launch SolO on an Atlas V 411
- Launch date: October, 2018

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Linking in-situ with remotesensing instruments



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- SIS: Supra-thermal lons spectrograph
- STEP: Supra-thermal Electrons and Protons
- EPT: Electron Proton Telescope
- HET: High Energy Telescope
- ICU: Instrument Control Unit

Electrons, protons, and heavy elements from few keV up to 400 MeV/n





METIS



- Observables
 - Coronal density
 - Coronal outflow velocity
- Where
 - Solar wind is accelerated
 - CMEs first propagate
 - Shock fronts accelerate particles

- Externally occulted coronograph
 - Annular FOV: 1.5°-2.9° (1.6 3.0 R_o @ 0.28 AU)
- Simultaneous imaging in two channels
 - Broadband polarized, visible light (580 640 nm)
 - Narrow band UV @ Lyman_{α} (121.6 ± 10 nm)
- Spatial resolution
 - ≥ 20" (vis. and UV phot. cont.); ≥ 4000 km @ 0.28 AU
- Temporal resolution: ≥ 1 min typically

The corona as a link to the heliosphere

(PI: E. Antonucci, Italy)

- Science goals
 - Links between the surface and higher layers
 - Mapping *B* in the photosphere (HR & FD to be extrapolated where necessary; in combination)
 - Energetics, dynamics, and fine structure of the magnetic field (... poles)
 - Mapping **B** and v_{LOS} (HR; in combination)
 - Probe the solar dynamo
 - Uninterrupted series of *B* and v_{LOS} (HR & FC; stand-alone)











Polarimetry

Spectroscopy

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HRT & FDT

Image stabilization system



Mapping

LCVR-based polarimeters



Tachography

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(PI: S.K. Solanki, Germany; co-PI: J.C. del Toro Iniesta, Spain)



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Mapping

Magnetometry

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(PI: S.K. Solanki, Germany; co-PI: J.C. del Toro Iniesta, Spain)

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Magnetometry

Mapping

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Imaging

Polarimetry

Spectroscopy

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LCVR-based polarimeters

Image stabilization system



Tachography

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LiNb O₃ etalon

HRT & FDT



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Imaging

Polarimetry

Spectroscopy

SPG

Mapping

Magnetometry

Tachography

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HRT &

Image

LCVR-b

LiNb C



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The measurement principle











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The measurement principle

> We infer the vector magnetic field and the LOS velocity by inverting the RTE on board by means of a special device

SO/PHI is
Differential imager
Diffraction limited
Wavelength tunable
Quasi monochromatic
Polarization sensitive
With sophisticated onboard capabilities



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S. Antiochos (NASA) R. Bruno (IAPS-INAF) P. Charbonneau (UMo) M. Collados (IAC) S. Cranmer (CU) L. Gizon (MPS) V. Hansteen (UiO) Y. Katsukawa (NAOJ) E. Kilpua (UH) A. Lagg (MPS) D. Lario (APL) V. Martinez Pillet (NSO) M. J. Owens (UR) S. Parenti (IAS) E. Priest (UStA) J.C. Raymond (CfA) K. Reeves (CfA) A. Rouillard (CNES) L. Rouppe van der Voort (UiO) T. Shimizu (ISAS) L. van Driel-Gesztelyi (MSSL) M. Velli (JPL) P. Young (NASA) F. Zucarello (UniCT)

7th Solar Orbiter Workshop Exploring the Solar Environs

http://spg.iaa.es/solo2017

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Granada, Spain, 3-7 April 2017

GOLERNO NENETERO DE ROMAN DE ROMONIAN COMPLETITAD CSIC

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Relative rotation

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esa Campaign duration

campaigns PHI	CB	Kate		alloc	requirements			86	
(HRT or FDT)	8	kbps	hrs	% of 53.14Gb		Operationa I feasibility	TM feasibility		
Near-surface rotation	n, meridia	nal circu							
surface flows seis1	14400 (2sets/ 8h)	2,9/ 3,09	720 (30 days)	14,21	30days at high latitude (limited time at 'high lat')	feasible outside baseline	TM feasible	EMP high latitude periods	
supergranulation tracking seis2	3600 (1/hr)	0,36/0, 39	720 (30 days)	1,78	30days at moderately high latitude (limited time at 'high lat')	feasible outside baseline	TM feasible	NMP high latitude periods	bad comms
helioseismology & solar cycle variations seis3	60 (1/min)	21,85/2 3,15	720 (30 days)	106,56	4 x 30d campaign, 2 years apart, at high latitudes (high latit, only in last 4 yrs)	partially, outside baseline	extra TM needed	around high lat windows, close to E	only partial cycle + data storage
Meridional circulation to 3m/s @75° sels4	TBD	TBD	2400 (+100 d)	TBD	>100d at high latitude (@75°)	unfeasible with current description	твс	TBD	TBD
Deep and large-scale solar dynamics									
MDI-like medium-l & stereoscopic helioseismology <i>seis5</i>	60 (1/min)	1,37/1, 27	>1 orbit long as possible	37,30	to be run continuously (esp. far side/high lat) Continuous FDT might conflict with RSWs	feasible outside baseline	TM feasible (**)	may fit outside RSWs	RSW TM reduction mid mission
Convection at high latitudes									
helioseismology selső OR	60 (1/min)	87,38/8 2,67	168 (7days)	99,46	7days at high latitudes	within RS baseline	TM limit (total alloc in 7d)	high latitude RSW, close to Earth	may need combination with sels3
local correlation tracking seis1 short	14400 (2sets/ 8h)	2,9/3,3 1	168 (7days)	3,32	7 days at high latitudes needs HRT & feature tracking?	within RS baseline	TM feasible	any high Iatitude RSW	
Deep convection and giant cells									
feature tracking 2x seis2	3600 (1/hr)	0,3 6/0, 39	1440 (60 days)	3,55	4 repetitions of 60days OR full orbit, high latitudes mainly (??)	outside baseline	TM feasible	TBD	
helioseismology seis5	60 (1/min)	1,37/1, 27	4032 (1 orbit)	37,30	to be run continuously (esp. far side/high lat)	feasible outside baseline	TM feasible (**)	= MDI-like campaign	RSW TM reduction mid mission



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esa **Campaign duration**

Helioseismology campaigns PHI	Caden ce	TM Rate	Duration	% orbital alloc	Orbital & operational requirements	Feasibility		Opportuni ties	Drawback
(HRT or FDT)	5	kbps	hrs	% of 53.14Gb		Operationa I feasibility	TM feasibility		
Active Regions and	sunspot	s							
AR flows and structure seis3 short	60	21,85/ 23,15	480 (20 days)	71,04	20 days of same AR: feature tracking @perihelion far side	within RS baseline	TM feasible	far side concat. RSWs	
sunspot oscillations seis7	60	262,1/ 231,5	48 (2 days)	85,25	2 days stereoscopic observations, feature tracking	within RS baseline	85% TM in 2d -> release TM in steps	RSW flying inwards	rest orbit TM low?
calibration far-side helioseismology seis5 short	60	1,37/1 ,74	48 (2 days)	0,44	to be repeated 5 times 2 days @ far side, AR on Earth side	within RS baseline	TM feasible	far side perihelion	
Physics of oscillation	ans								
effect granulation on oscillations seis8	60	262,1 231,5	24	42,63	Stereoscopy + (clipped?) 'raw' data download	within RS baseline	TM feasible (**)	within RS window	40% TM in 1 day
two components of velocity (v) seis3 short	60	21,85/ 20	240 (10days)	35,52	observe several lifetimes of super-granulation	within RS baseline	TM feasible (**)	full RSW	
magnetic oscillations (MHD waves) <i>seis</i> 9	60 (B at highest	1748/ 1157	24	284,17	stereoscopic observ of AR: HRT & feature tracking?	within RSbaseline	extra TM needed	last RSW before underrun	Space needed in PHI buffer
Low resolution obse	ervatione	1							
LOI-like observations (sun-as-a-star)	60	0,088 0,003	4032 (1 orbit)	2,40	higher latitudes, continuous observations	outside baseline	TM feasible	TBD	
Shape of the sun seis11	3600 (1set/ hr)	0,083	12 (1 roll)	0,0068	spacecraft rolls in 12 discrete steps	твс	TM feasible	TBD	

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